

JEM/SMILES L2 data processing system on ISAS/JAXA

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and SMILES mission team^{*2,5}

1: FUJITSU FIP, 2: ISAS/JAXA, 3:TOME R&D, 4:RISH, 5: NICT

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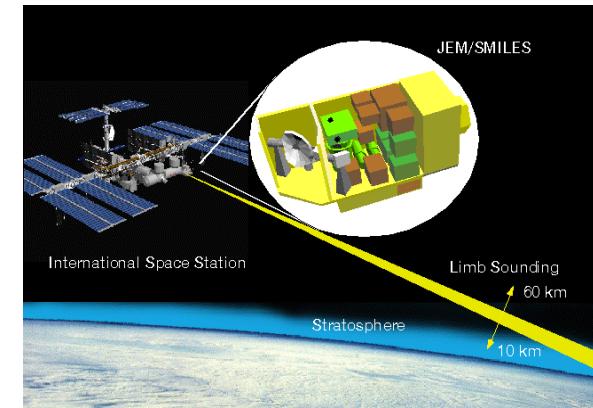
- Outline of the SMILES Level 2 data processing system (DPS-L2) in ISAS/JAXA
 - Retrieval algorithm of the DPS-L2
 - Study of the observation capability for target species.
 - Improvements of the level 2 data.
-
- ◆ Recent status of the Level 2 products is NEXT talk.

Outline of JEM/SMILES

- The Superconducting Submillimeter-wave Limb-Emission Sounder (SMILES) had been launched and aboard the Japanese Experiment Module (JEM) of the International Space Station (ISS) in **Sep, 2009**.
- The SMILES carries 4K cooled **Superconductor-Insulator-Superconductor (SIS) mixers** to demonstrate a highly sensitive instrument for submillimeter limb-emission sounding.
 - ◆ Global test observation : Oct.12.
 - ◆ Operational observation : Nov. 6.

Standard products

O₃, HCl, ClO, HNO₃, CH₃CN, HO₂, HOCl, BrO, O₃ isotopes
(in the middle atmosphere)

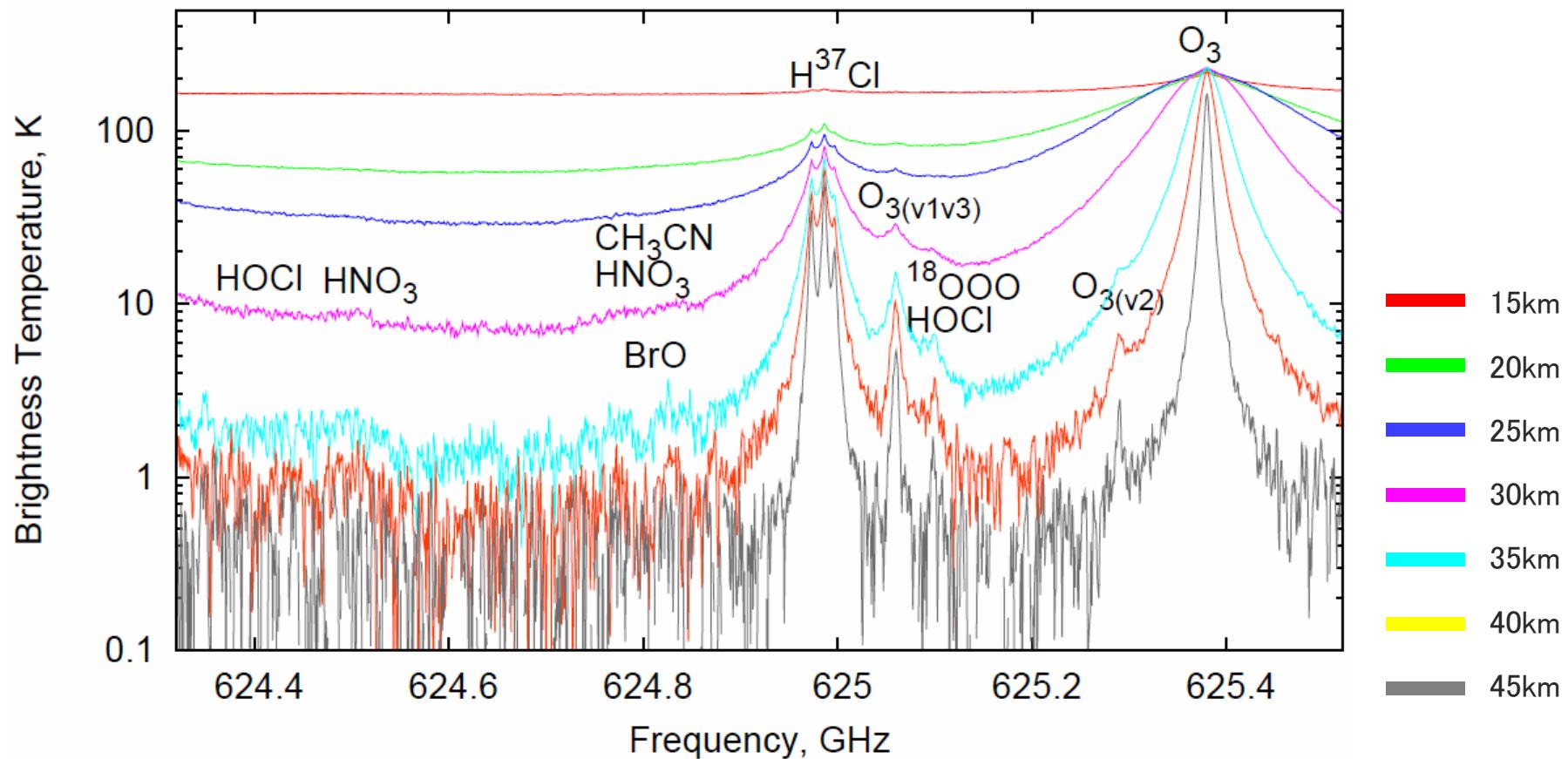


JEM/SMILES Specifications

Type	Parameters
Frequency bands	Band A (624.32 – 625.52 GHz) Band B (625.12 – 626.32 GHz) Band C (649.12 – 650.32 GHz)
System noise temperature	Less than 500 K
Instrumental height resolution	3.5 - 4.1 km (nominal)
Frequency resolution	1.8 MHz (FWHM)
Channel separation	0.8 MHz /channel
Observation altitude range	10 - 60 km (nominal)
Global coverage	38S - 65N (nominal)
Observation azimuth angle	-10 – 95 degree (0=north)
Processing time	53 s / scan
Integration time	0.5 s for each observation point

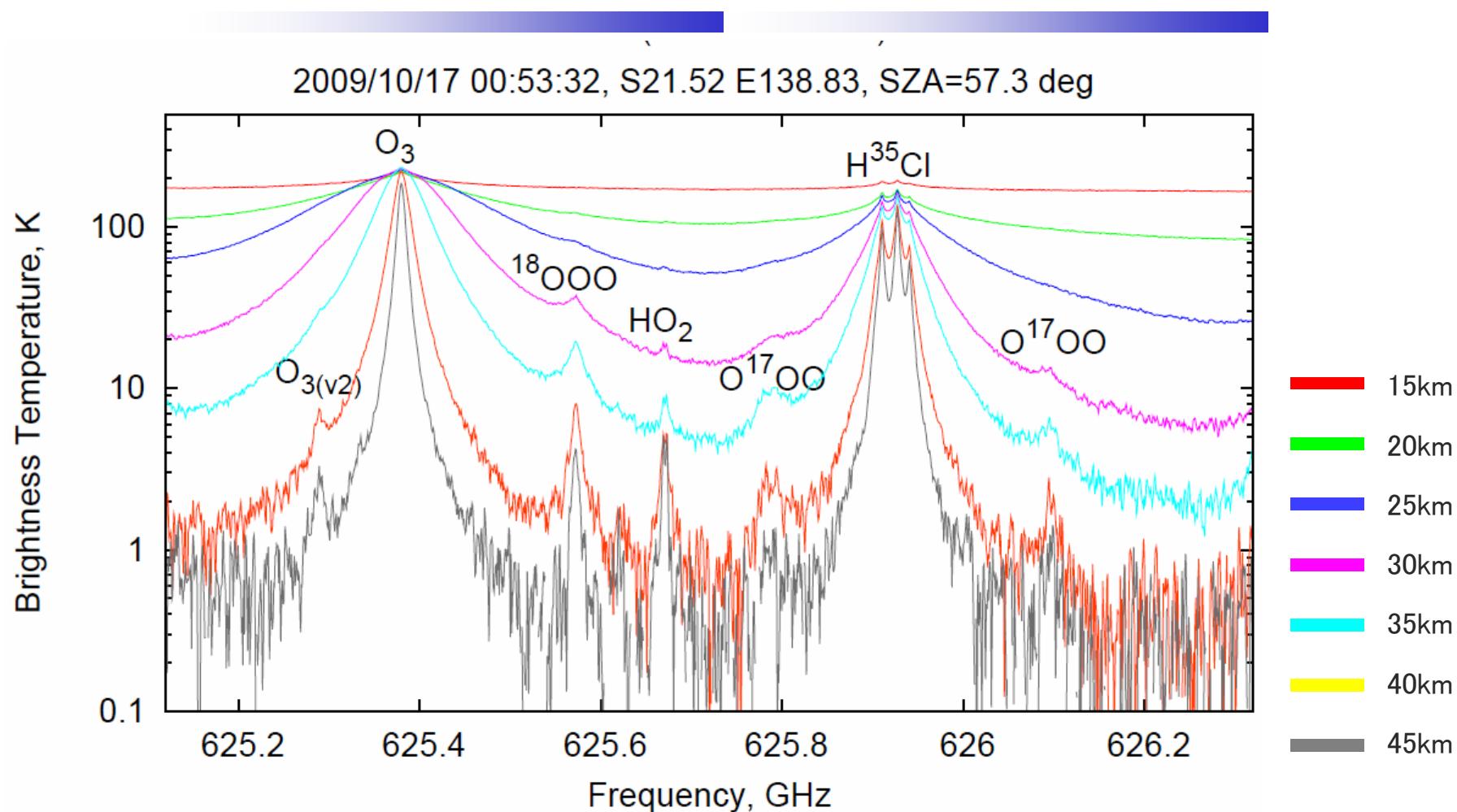
Observation band A (091012000062)

2009/10/12 03:22:14, N23.30 W173.83, SZA=55.8 deg



- Strong lines:: O₃ and HCl, Weak lines:: HOCl, HNO₃, CH₃CN and BrO (band A only : CH₃CN, HOCl)
- BrO line is superimposed on HNO₃ lines. Other major HNO₃ line is only one and it is not clearly. So, BrO retrieval is difficult problem.

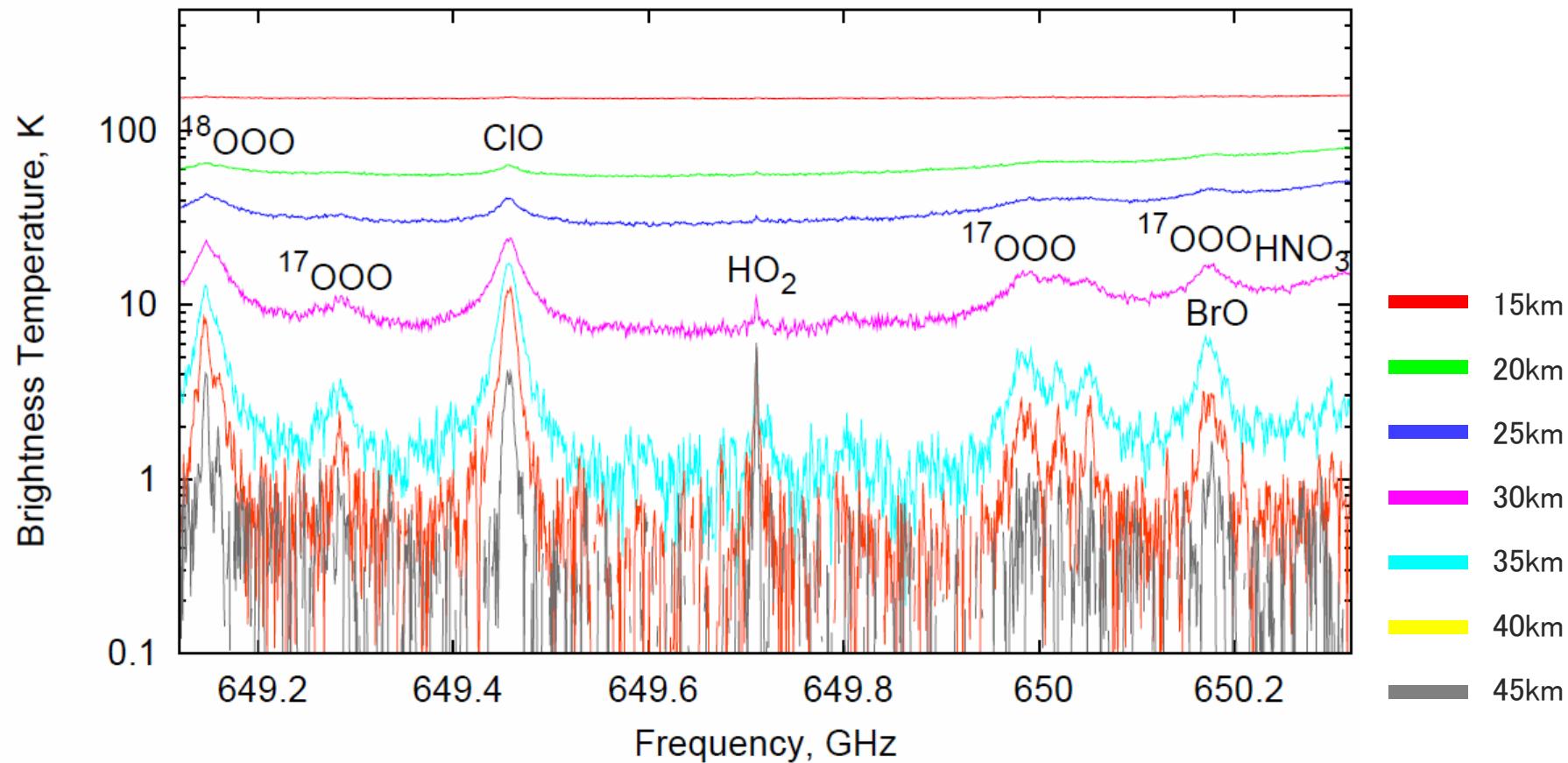
Observation band B (091018000060)



- Strong lines:: O₃ and HCl, Weak lines:: O₃-isotope and HO₂ (band B only : O¹⁷OO)
- HCl signal in stratosphere is larger than signal in band A.
- Major lines are relatively isolated

Observation band C (091012000062)

2009/10/12 03:22:14, N23.30 W173.83, SZA=55.8 deg



- Strong lines:: CIO, O₃-isotope Weak lines: HNO₃ HO₂, BrO. (band C only: CIO, ^{17}OOO)
- BrO line is superimposed on ^{17}OOO lines. But ^{17}OOO has other 7 strong lines which are isolated from other molecular's lines. So, Band-C is relatively appropriate to retrieve BrO.

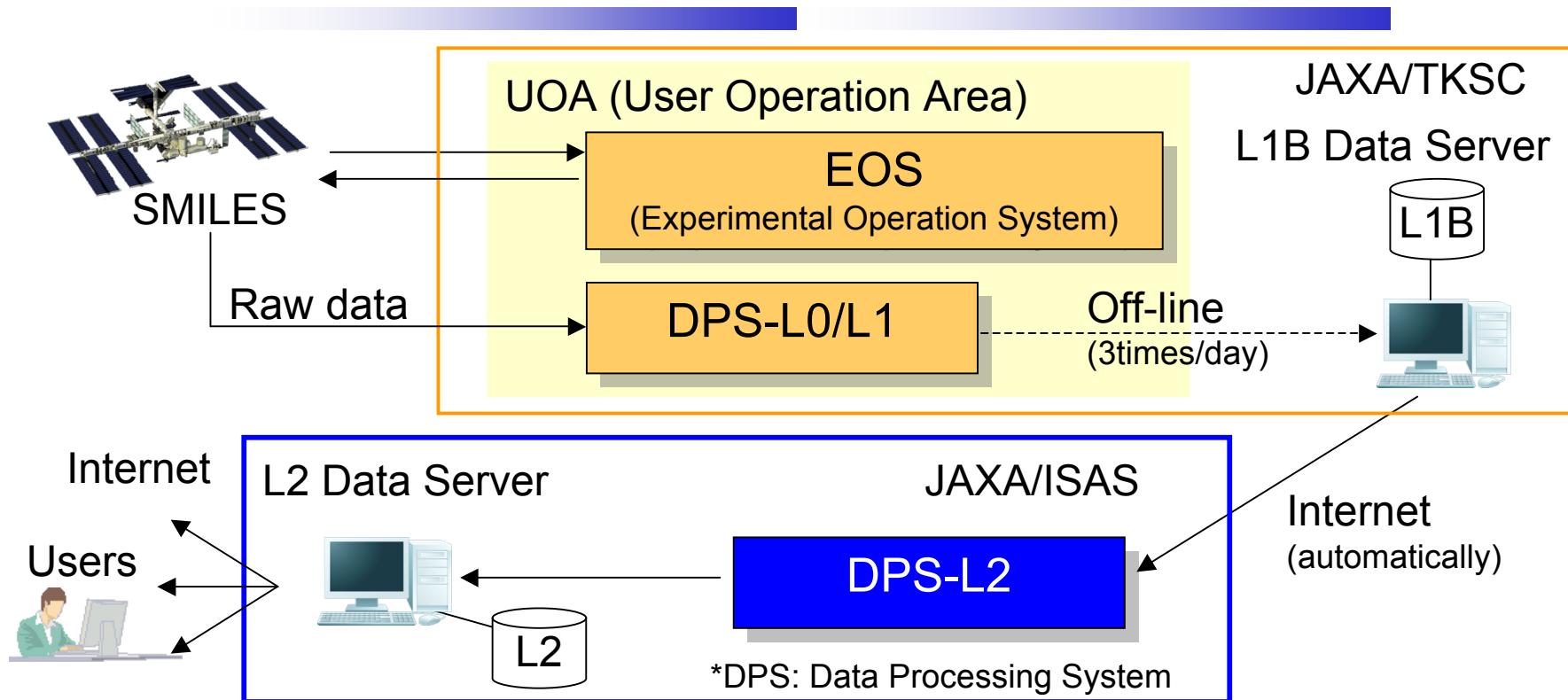
JEM/SMILES Data Sets

Data Type	Description
RAW	Unprocessed mission data at binary packets
Level 0	Reconstructed, unprocessed mission data at binary packets
Level 1b	Calibrated instrument radiances and related data
Level 2	Derived geophysical variables at the same resolution and location as the Level 1 source data
Level 3	Variables mapped on uniform space-time grid scales, usually with some completeness and consistency



Processing

JEM/SMILES Data Flow



- Downlinked raw data from the SMILES will be received by the DPS-L0/L1 at User Operation Area (UOA) on Tsukuba Space Center (TKSC).
- The DPS-L0/L1 processes the raw data consisting of house keeping (HK) data and mission data to brightness temperature (level 1B data) in near-real-time.
- The DPS-L2 produces the vertical profiles of target species called level 2 data in near real time and upload L2 product dataset 5 day after observation (TDB).
- The DPS-L2 distributes the level 2 product to users by a Web server.

Basic Retrieval Algorithm

■ Optimal Estimation Method (OEM)

- Observation vector:
 \mathbf{x} :true, $\boldsymbol{\epsilon}$:observation noise , \mathbf{f} :Forward Model
- OEM: *deriving the results which minimize χ^2*

$$\chi^2 = [\mathbf{y} - \mathbf{f}(\mathbf{x})]^T \mathbf{S}_y^{-1} [\mathbf{y} - \mathbf{f}(\mathbf{x})] + [\mathbf{x}_a]^T \mathbf{S}_a^{-1} [\mathbf{x}_a - \mathbf{x}_a]$$

\mathbf{x}_a :a priori, \mathbf{S}_a :covariance of a priori, \mathbf{S}_y covariance of observation noise

■ Non-linear case (Levenberg-Marquardt Method)

$$\mathbf{x}_{i+1} = \mathbf{x}_i + (\mathbf{S}_a^{-1} + \mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i + \gamma \mathbf{S}_a)^{-1} \left\{ \mathbf{K}_i^T \mathbf{S}_y^{-1} [\mathbf{y} - \mathbf{F}(\mathbf{x}_i)] - \mathbf{S}_a^{-1} [\mathbf{x}_i - \mathbf{x}_a] \right\}$$

\mathbf{K}_i :Weighting function, γ :Levenberg-Marquardt parameter

References for the SMILES L2 data processing

- C. Takahashi, S. Ochiai, M. Suzuki, Operational Retrieval Algorithms for JEM/SMILES Level 2 Data Processing System, JQSRT (in press)
- K. Imai, M. Suzuki, C. Takahashi, Evaluation of the Voigt algorithms for the ISS/JEM/SMILES L2 data processing system, ASR (in press)

Algorithm for noisy products

- Retrieval profiles of noisy products may include a priori information.
- To avoid the bias from a priori, we retrieve the multi-scan data simultaneously [Livesey,2004].
 - The observation data \mathbf{y}_i ($i=1 \sim N$), the weighting function \mathbf{K}_i , than reference spectra \mathbf{f}_i , and the covariance matrix of the measurements \mathbf{S}_{yi} are represented by :

$$\mathbf{y} = \begin{pmatrix} \mathbf{y}_1 \\ \mathbf{y}_2 \\ \vdots \\ \mathbf{y}_N \end{pmatrix}, \quad \mathbf{f} = \begin{pmatrix} \mathbf{f}_{i1} \\ \mathbf{f}_2 \\ \vdots \\ \mathbf{f}_N \end{pmatrix}, \quad \mathbf{K} = \begin{pmatrix} \mathbf{K}_1 \\ \mathbf{K}_2 \\ \vdots \\ \mathbf{K}_N \end{pmatrix}, \quad \mathbf{S}_y = \begin{pmatrix} \mathbf{S}_{y1} & \mathbf{0} & \cdots & \mathbf{0} \\ \mathbf{0} & \mathbf{S}_{y1} & \mathbf{0} & \vdots \\ \vdots & \mathbf{0} & \ddots & \mathbf{0} \\ \mathbf{0} & \cdots & \mathbf{0} & \mathbf{S}_{y1} \end{pmatrix}$$

- However, these matrixes are too large (ex. \mathbf{K}_i size: 1500x30x30), we calculate $\mathbf{K}_i^T \mathbf{S}_{yi} \mathbf{K}_i$ (size: 30x30) and $\mathbf{K}_i^T \mathbf{S}_{yi} (\mathbf{y}_i - \mathbf{f}_i)$ (size:30) for each scan and save to reduce the load of the system.

$$\hat{\mathbf{x}} = \mathbf{a} + \left[\mathbf{S}_a^{-1} + \sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i \right] \left[\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} [\mathbf{y}_i - \mathbf{f}_i(\mathbf{x}_0, \mathbf{b})] - \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i [\mathbf{a} - \mathbf{x}_0] \right]$$
$$\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i = (\mathbf{K}_1^T \mathbf{S}_{y1}^{-1} \mathbf{K}_1 + \mathbf{K}_2^T \mathbf{S}_{y2}^{-1} \mathbf{K}_2 + \cdots + \mathbf{K}_N^T \mathbf{S}_{yN}^{-1} \mathbf{K}_N)$$
$$\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} [\mathbf{y}_i - \mathbf{f}_i(\mathbf{x}_0, \mathbf{b})] = (\mathbf{K}_1^T \mathbf{S}_{y1}^{-1} [\mathbf{y}_1 - \mathbf{f}_1(\mathbf{x}_0, \mathbf{b})] + \cdots + \mathbf{K}_N^T \mathbf{S}_{yN}^{-1} [\mathbf{y}_N - \mathbf{f}_N(\mathbf{x}_0, \mathbf{b})])$$

Reference : N J. Livesey and W. Van Snyder, EOS MLS Retrieval Processes Algorithm Theoretical Basis, 2004

Error Analysis

■ Retrieval Precision

- *Measurement error* : decided by the system noise

$$\mathbf{S}_m = (\mathbf{S}_a^{-1} + \mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i)^{-1} \mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i (\mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i + \mathbf{S}_a^{-1})^{-1}$$

- *Smoothing error* : decided by the atmospheric conditions

$$\mathbf{S}_n = (\mathbf{S}_a^{-1} + \mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i)^{-1} \mathbf{S}_a^{-1} (\mathbf{K}_i^T \mathbf{S}_y^{-1} \mathbf{K}_i + \mathbf{S}_a^{-1})^{-1}$$

■ Forward Model parameter error

$$\mathbf{S}_f = \mathbf{G}_y \mathbf{K}_b \mathbf{S}_b \mathbf{K}_b^T \mathbf{G}_y^T$$

- *Instrumental parameter error*
- *Spectroscopic parameter error*
- *Approximation and implementation error*
- etc.

- Forward Model parameter error << Retrieval Precision
→ High accuracy algorithm is required.

Doppler shift & Instrument Functions

■ Doppler shift

- Velocity of the ISS : 8 km/s
- Rotation of the earth : 460 m/s (on the equator)
- Wind : ≤ 100 m/s. (Use GMAO)

■ FOV Convolution :

Considering the effects from tilt of ISS

$$T_A(\nu, z_0) = \int_{z_{min}}^{z_{max}} P(z, z_0) \cdot T_p(\nu, z) dz,$$

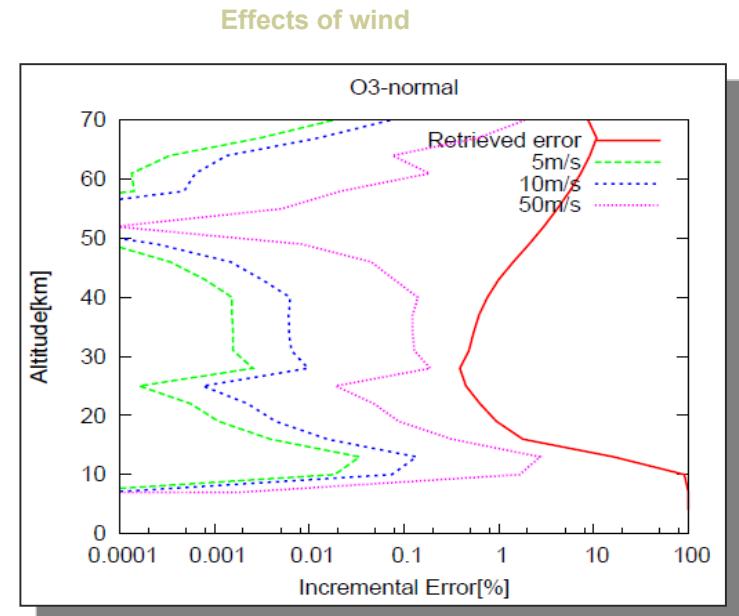
■ Fold Sidebands (Single Sideband Separator)

$$T_{mix:i}(\nu, z_0) = \begin{bmatrix} K_{i,a}^{LSB}(\nu_{LO} - \nu_{if}, z_0) \\ K_{i,c}^{LSB}(\nu_{LO} - \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} - \nu_{if}, z_0) \\ T_c(\nu_{LO} - \nu_{if}) \end{bmatrix} + \begin{bmatrix} K_{i,a}^{USB}(\nu_{LO} + \nu_{if}) \\ K_{i,c}^{USB}(\nu_{LO} + \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} + \nu_{if}, z_0) \\ T_c(\nu_{LO} + \nu_{if}) \end{bmatrix}$$

$$K_{i,j}^{LSB,USB}(\nu, T) = \frac{1 + \alpha(T)^2 + 2\alpha(T)\cos\left(\frac{m\pi\nu}{\nu_0(T)}\right)}{4}.$$

■ Channel Average: AOS Response function approximates as Gaussians. And Gaussian parameters depend on channel number.

$$T_{AOS(l)}(\nu_j, z_0) = \frac{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) \cdot T_{mix(k)}(\nu, z_0) d\nu}{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) d\nu}. \quad H_{AOS(l)}(\nu - \nu_j) = \sum_{j=1}^{N_l} \frac{A_{i,j}}{w_{i,j}\sqrt{\pi/2}} \cdot \exp\left(-2\frac{(\nu - \nu_j - xc_{i,j})^2}{w_{i,j}^2}\right).$$



A Priori / first guess

	Object	Reference data
Tracer constituents	O ₃	LBLRTM ^{*1} , Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	O ₃ isotopes	<i>O₃ references with isotopic abundances for HITRAN</i>
	HCl	LBLRTM ^{*1} , Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	CIO	LBLRTM ^{*1} , Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	CH ₃ CN	UARS/MLS ^{*4} (, Aura/MLS ^{*2})
	HOCl	LBLRTM ^{*1} , Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	HNO ₃	LBLRTM ^{*1} , Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	HO ₂	Aura/MLS ^{*2} , CCSR/NIES ^{*3}
	BrO	Aura/MLS ^{*2} , CCSR/NIES ^{*3}
Dynamics	Temperature H ₂ O Wind Pressure	NASA/GMAO ^{*5}

*1: Reference profiles used in Line-by-Line Radiative Transfer Model

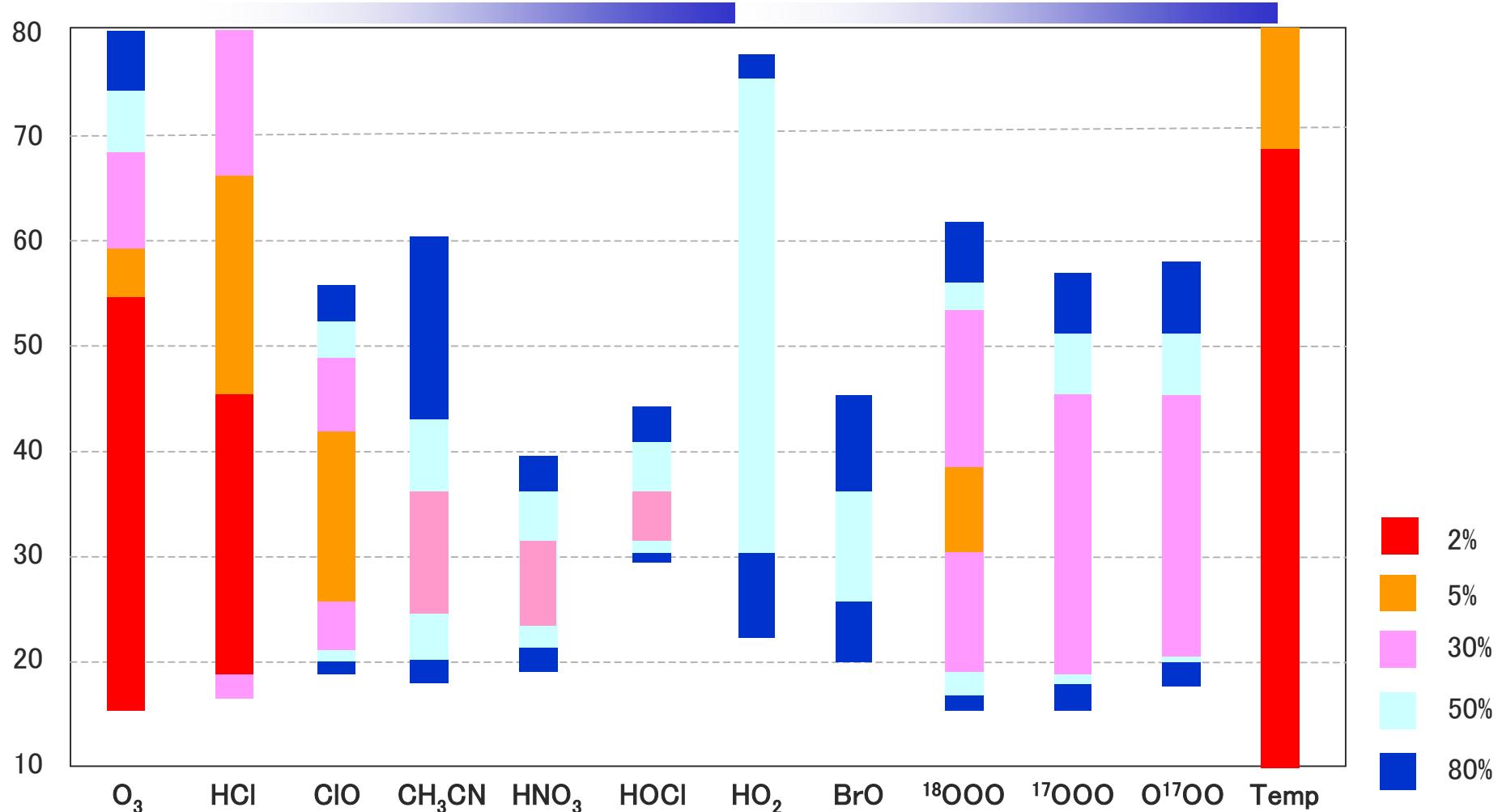
*2: Monthly climatology based on measurements by EOAS-Aura/MLS

*3: Monthly climatology based on the CCMVal-REF2 run for 2001-2010 by CCSR/NIES CCM

*4: CH₃CN reference profiles based on measurements by UARS/MLS

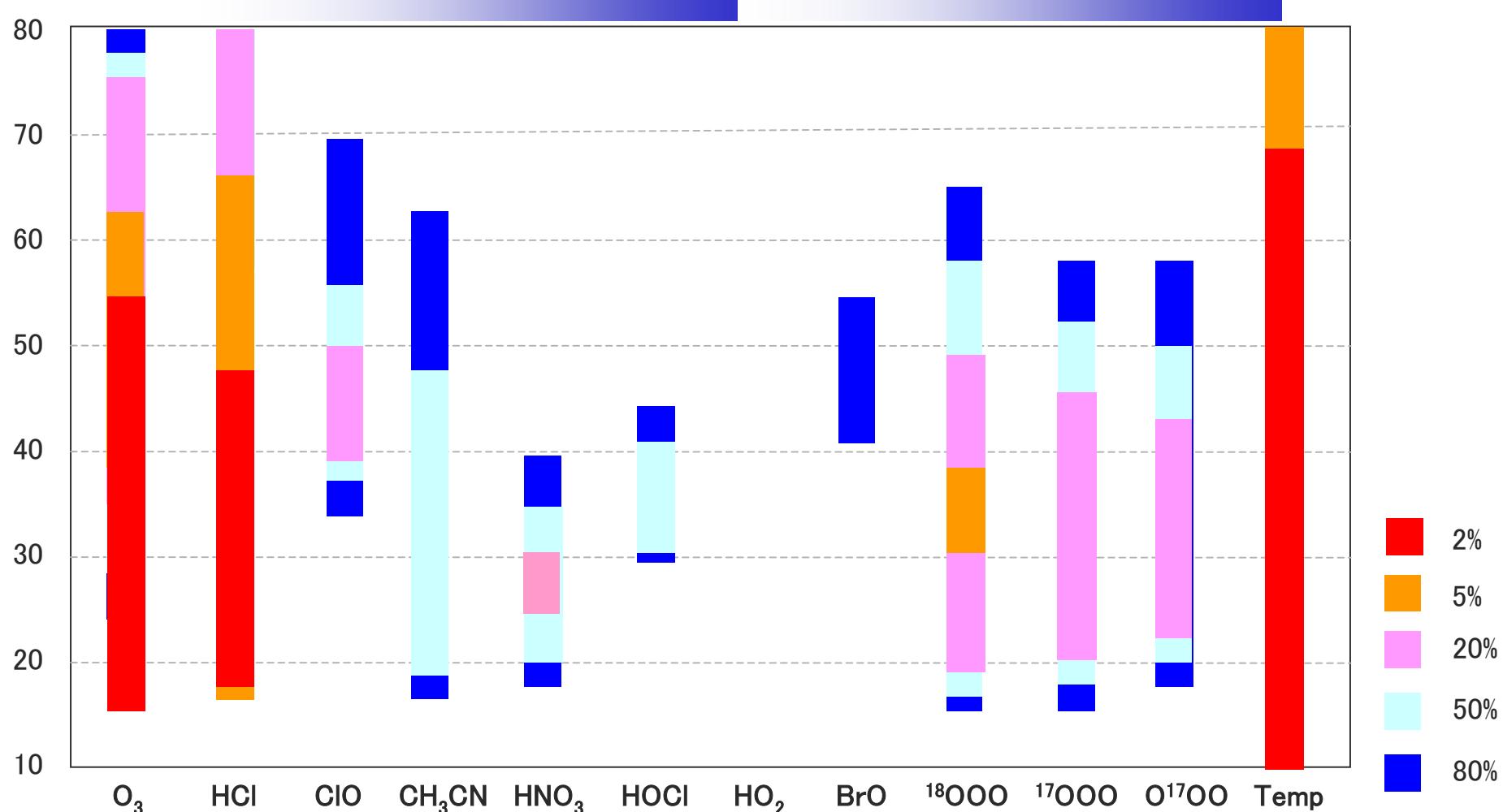
*5: Near-realtime analyses produced by NASA/GMAO's GEOS-5 DAS

Retrieval Precision : Day (LT12:00)



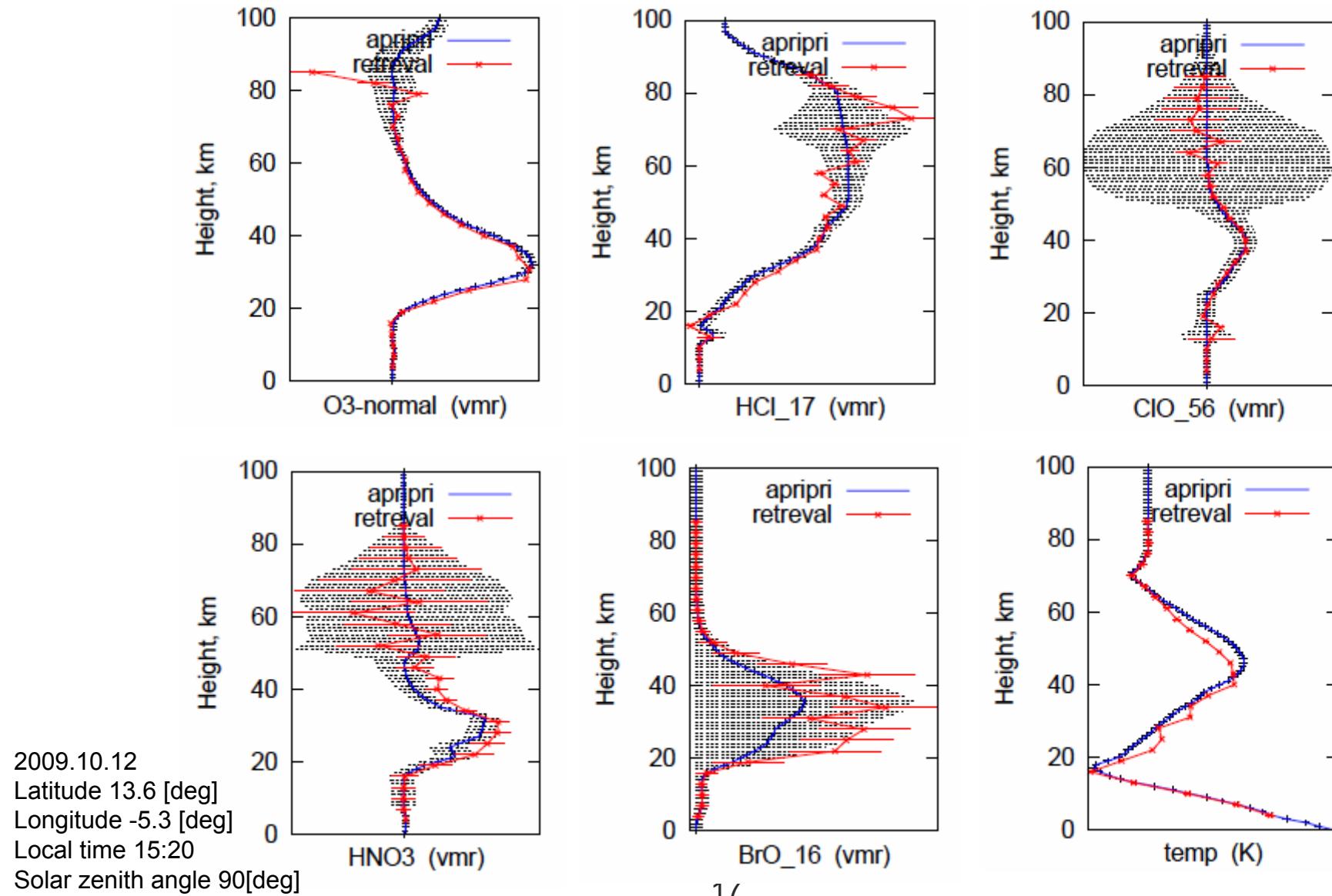
- a priori profiles HOCl, BrO: CCSR/NIES, CH_3CH : UARS/MLS, others: Aura/MLS (40N/March)
- Standard deviation of a priori : 100 %
- Measurement altitude : 0-80km, 2km, Retrieval Altitude: 4-85km, 3km

Retrieval Precision : Night (LT0:00)



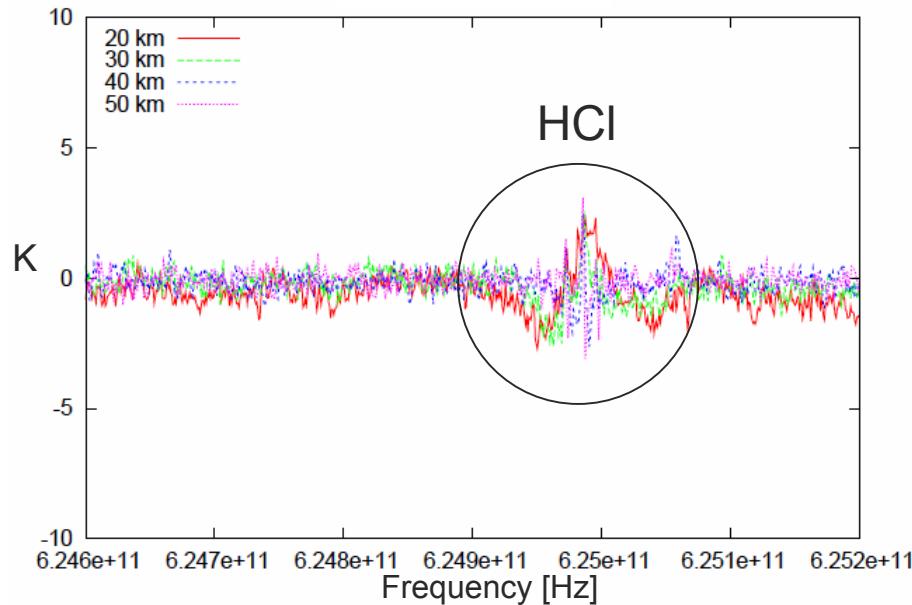
- a priori profiles HOCl, BrO: CCSR/NIES, CH₃CH: UARS/MLS, others: Aura/MLS (40N/March)
- Standard deviation of a priori : 100 %
- Measurement altitude : 0-80km, 2km, Retrieval Altitude: 4-85km, 3km

Retrieval Results

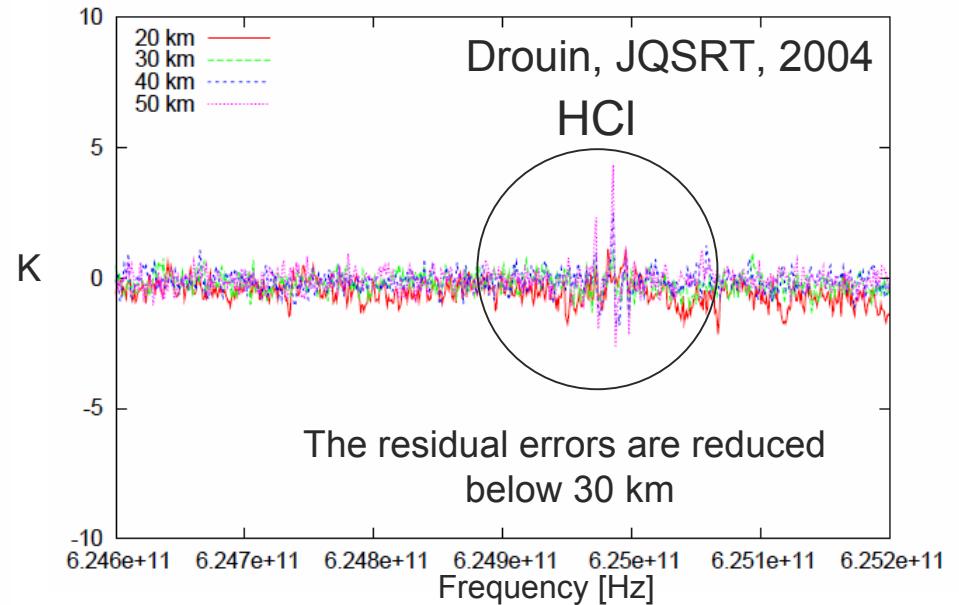


Spectroscopic parameter error (HCl)

Before the Correction
Residual error



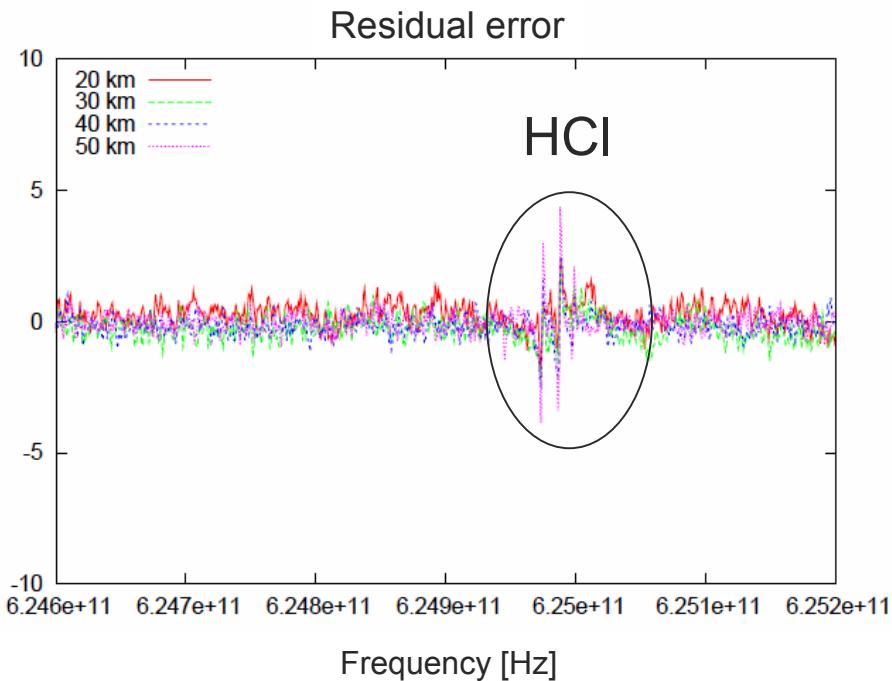
After the Correction
Residual error



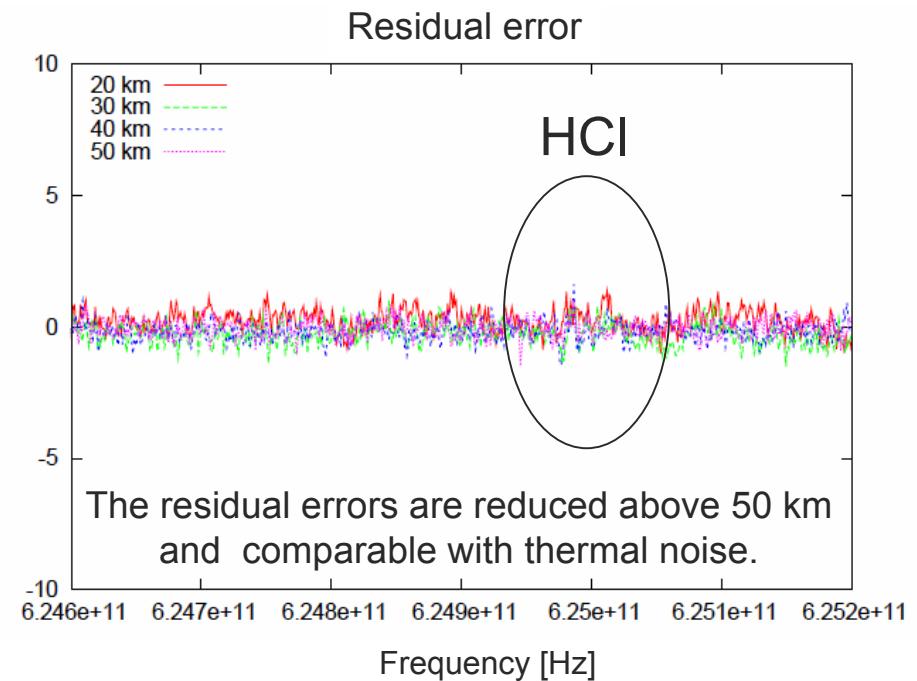
- Residual errors are 3 K around HCl lines at several heights. Below 30km, these errors are large around HCl lines, not other lines.
- Change HCl line parameters
 - include pressure shift effect
 - change temperature dependency of pressure broadening parameter

Doppler Shift by Earth's Rotation Rate

Before the Correction



After the Correction



- SMILES antenna viewing direction is from north-northwest to east.
These errors occur when direction is north-northwest (azimuth < 0 deg)
- Effect of Earth rotation rate is estimated by L1B data processing. So we request correction.

Summary

- The DPS-L2 is working as expected. The observation data can be retrieved with minimal loss.
- There are still some systematic errors in retrieval results. We will continue the following estimation and improve the level 2 data.
 - Accuracy of determining tangent heights
 - Error source of the frequency shift
(Frequency calibration, Wind, etc)
 - Line shape for other molecules
 - etc.

Algorithm for noisy products

- To avoid the bias from a priori, we retrieve the multi-scan data simultaneously [Livesey,2004]. i.e. The observation data \mathbf{y}_i ($i=1 \sim N$), the weighting function \mathbf{K}_i , than reference spectra \mathbf{f}_i , and the covariance matrix of the measurements \mathbf{S}_{yi} are represented by :

$$\mathbf{y} = \begin{pmatrix} \mathbf{y}_1 \\ \mathbf{y}_2 \\ \vdots \\ \mathbf{y}_N \end{pmatrix}, \quad \mathbf{f} = \begin{pmatrix} \mathbf{f}_{i1} \\ \mathbf{f}_2 \\ \vdots \\ \mathbf{f}_N \end{pmatrix}, \quad \mathbf{K} = \begin{pmatrix} \mathbf{K}_1 \\ \mathbf{K}_2 \\ \vdots \\ \mathbf{K}_N \end{pmatrix}, \quad \mathbf{S}_y = \begin{pmatrix} \mathbf{S}_{y1} & \mathbf{0} & \cdots & \mathbf{0} \\ \mathbf{0} & \mathbf{S}_{y1} & \mathbf{0} & \vdots \\ \vdots & \mathbf{0} & \ddots & \mathbf{0} \\ \mathbf{0} & \cdots & \mathbf{0} & \mathbf{S}_{y1} \end{pmatrix}$$

- However, these matrixes are too large (ex. K size: 1500x30x30), we calculate $\mathbf{K}_i^T \mathbf{S}_{yi} \mathbf{K}_i$ (size30x30) and $\mathbf{K}_i^T \mathbf{S}_{yi} (\mathbf{y}_i - \mathbf{f}_i)$ (size:30) for each scan and save to reduce the load of the system.

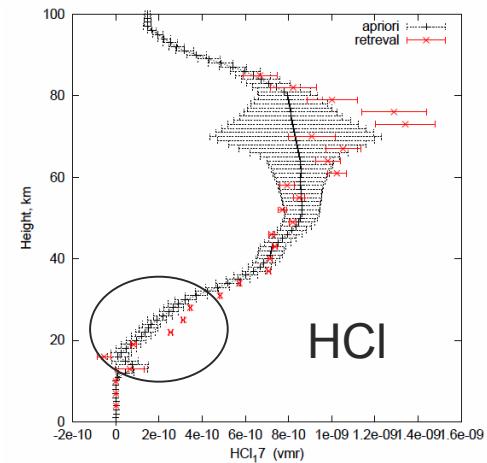
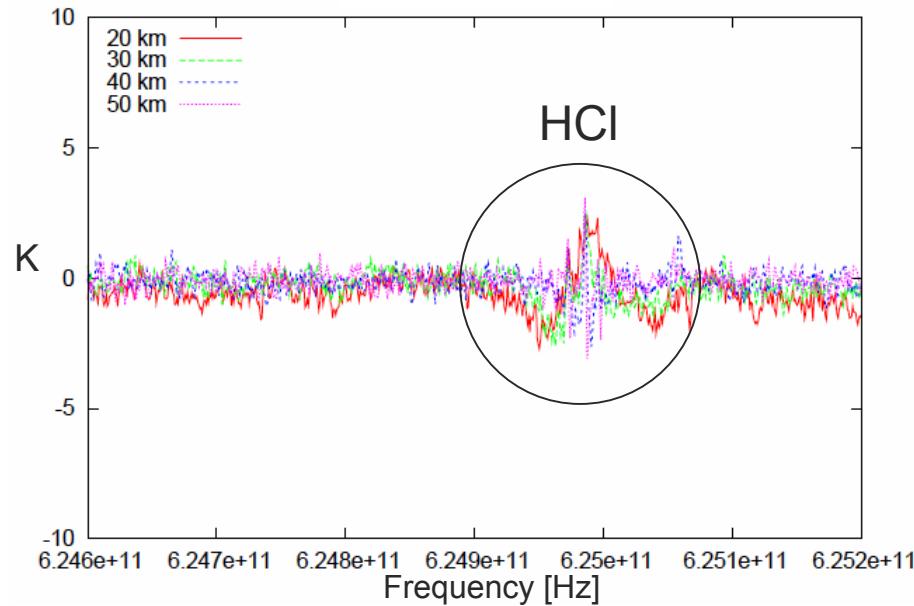
$$\hat{\mathbf{x}} = \mathbf{a} + \left[\mathbf{S}_a^{-1} + \sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i \right] \left[\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} [\mathbf{y}_i - \mathbf{f}_i(\mathbf{x}_0, \mathbf{b})] - \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i [\mathbf{a} - \mathbf{x}_0] \right]$$

$$\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} \mathbf{K}_i = \left(\mathbf{K}_1^T \mathbf{S}_{y1}^{-1} \mathbf{K}_1 + \mathbf{K}_2^T \mathbf{S}_{y2}^{-1} \mathbf{K}_2 + \cdots + \mathbf{K}_N^T \mathbf{S}_{yN}^{-1} \mathbf{K}_N \right)$$

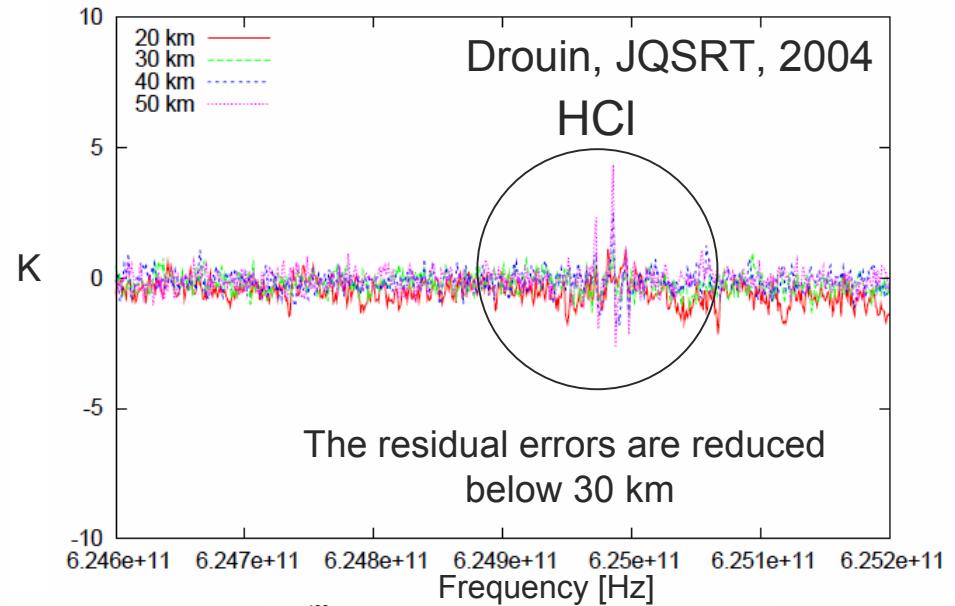
$$\sum \mathbf{K}_i^T \mathbf{S}_{yi}^{-1} [\mathbf{y}_i - \mathbf{f}_i(\mathbf{x}_0, \mathbf{b})] = \left(\mathbf{K}_1^T \mathbf{S}_{y1}^{-1} [\mathbf{y}_1 - \mathbf{f}_1(\mathbf{x}_0, \mathbf{b})] + \cdots + \mathbf{K}_N^T \mathbf{S}_{yN}^{-1} [\mathbf{y}_N - \mathbf{f}_N(\mathbf{x}_0, \mathbf{b})] \right)$$

Pressure Broadening Parameters (HCl)

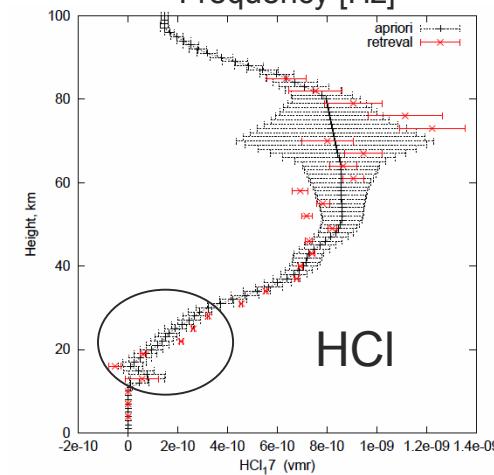
Before the Correction
Residual error



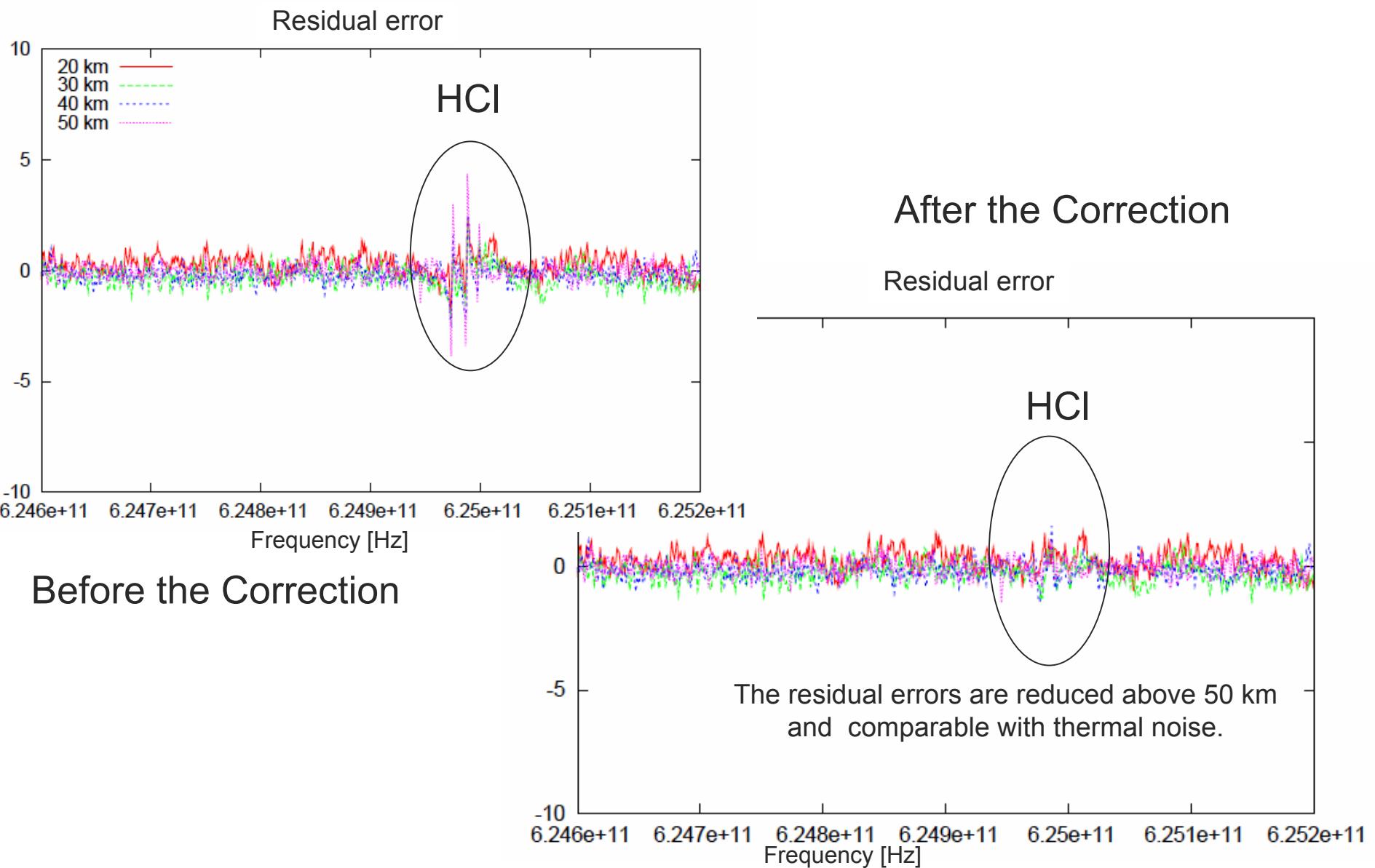
After the Correction
Residual error



The residual errors are reduced below 30 km



Doppler Shift by Earth's Rotation Rate



Doppler shift & Instrument Functions

Doppler shift

- Velocity of the ISS : 8 km/s
- Rotation of the earth : 460 m/s (on the equator)
- Wind : ≤ 100 m/s. (Use GMAO)

FOV Convolution :

Considering the effects from tilt of ISS

$$T_A(\nu, z_0) = \int_{z_{min}}^{z_{max}} P(z, z_0) \cdot T_p(\nu, z) dz,$$

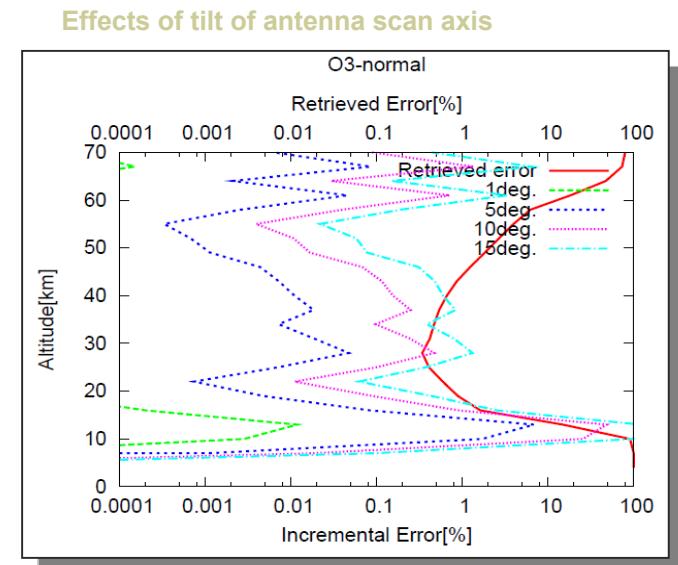
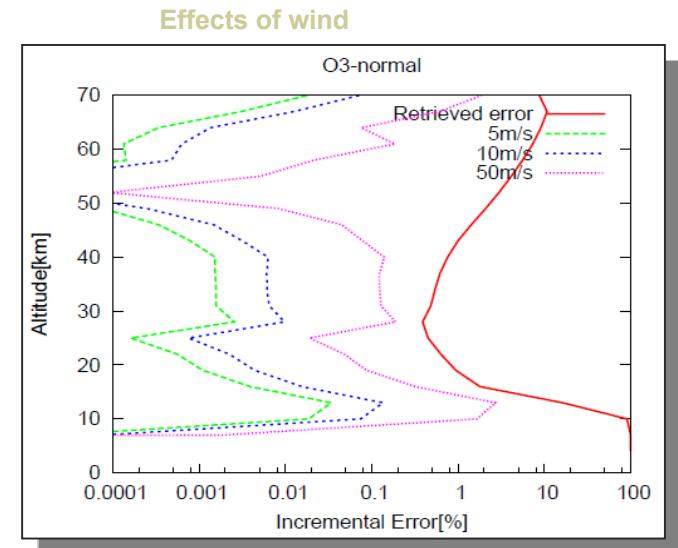
Fold Sidebands (Single Sideband Separator)

$$\begin{aligned} T_{mix:i}(\nu, z_0) &= \begin{bmatrix} K_{i,a}^{LSB}(\nu_{LO} - \nu_{if}, z_0) \\ K_{i,c}^{LSB}(\nu_{LO} - \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} - \nu_{if}, z_0) \\ T_c(\nu_{LO} - \nu_{if}) \end{bmatrix} \\ &+ \begin{bmatrix} K_{i,a}^{USB}(\nu_{LO} + \nu_{if}) \\ K_{i,c}^{USB}(\nu_{LO} + \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} + \nu_{if}, z_0) \\ T_c(\nu_{LO} + \nu_{if}) \end{bmatrix} \\ K_{i,j}^{LSB,USB}(\nu, T) &= \frac{1 + \alpha(T)^2 + 2\alpha(T)\cos\left(\frac{m\pi\nu}{v_0(T)}\right)}{4}. \end{aligned}$$

Channel Average: Considering channel dependence

$$T_{AOS(l)}(\nu_j, z_0) = \frac{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) \cdot T_{mix(k)}(\nu, z_0) d\nu}{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) d\nu}.$$

$$H_{AOS(l)}(\nu - \nu_j) = \sum_{j=1}^{N_j} \frac{A_{i,j}}{w_{i,j}\sqrt{\pi/2}} \cdot \exp\left(-2\frac{(\nu - \nu_j - xc_{i,j})^2}{w_{i,j}^2}\right).$$



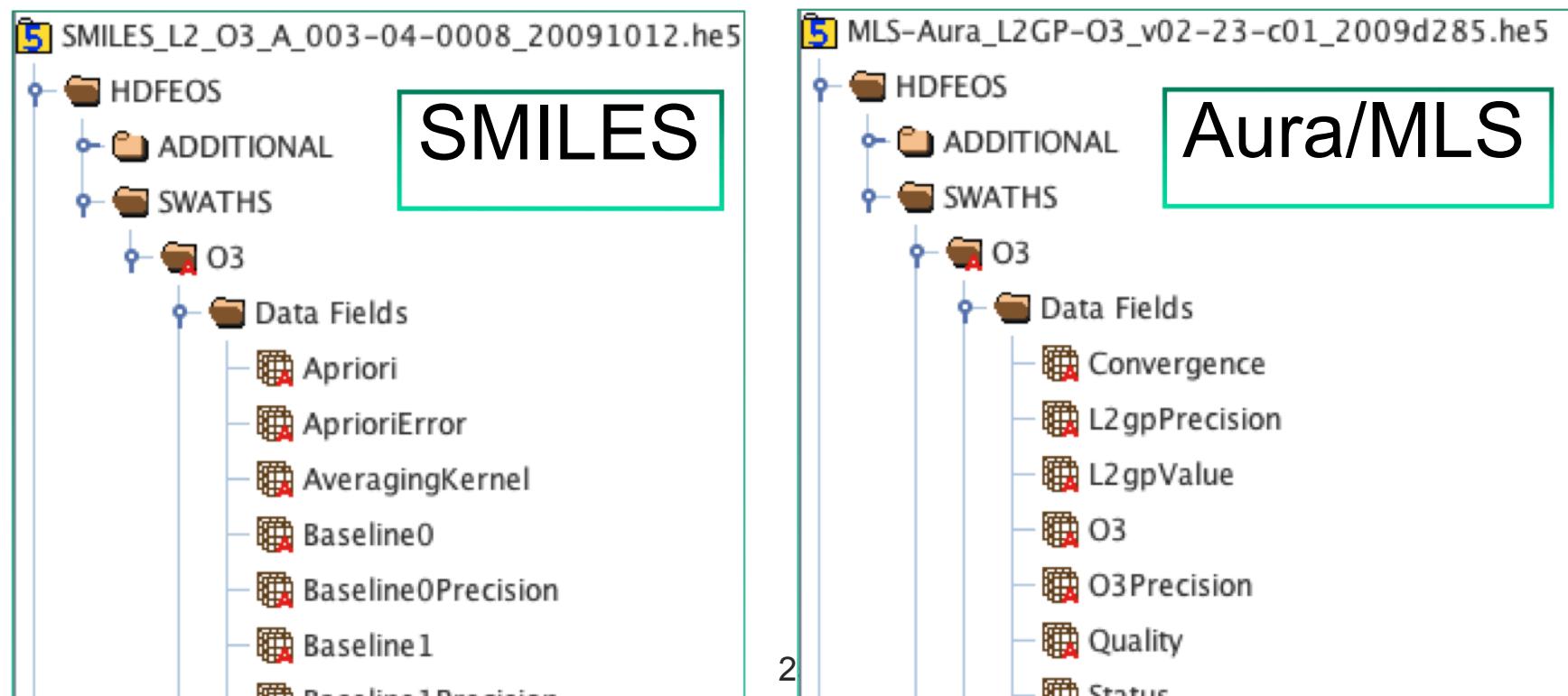
L2 Product dataset

Data: profile, precision, location, time, status....

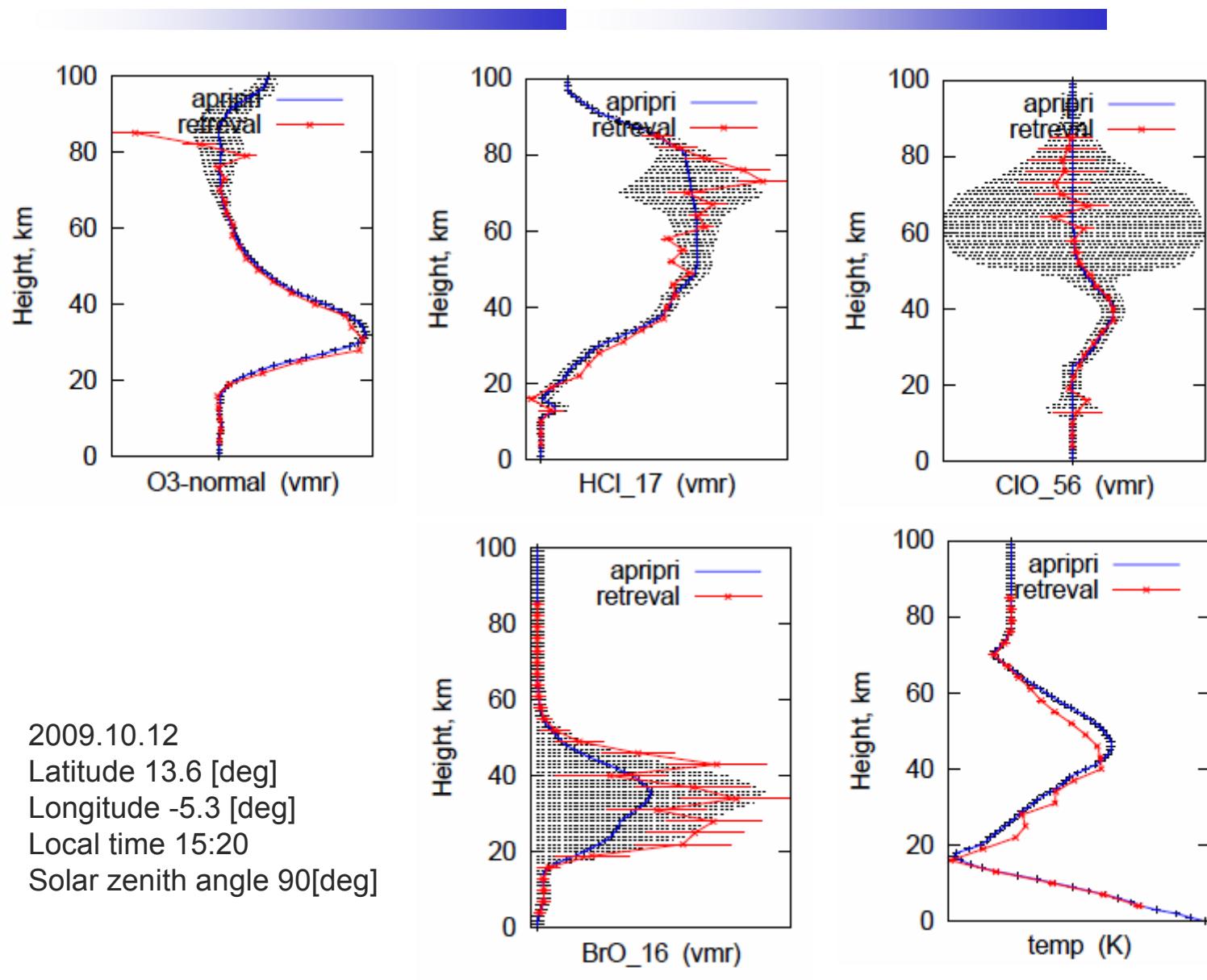
1 file / day / molecule (11MB / day)

Data format: EOS-HDF (version 5)

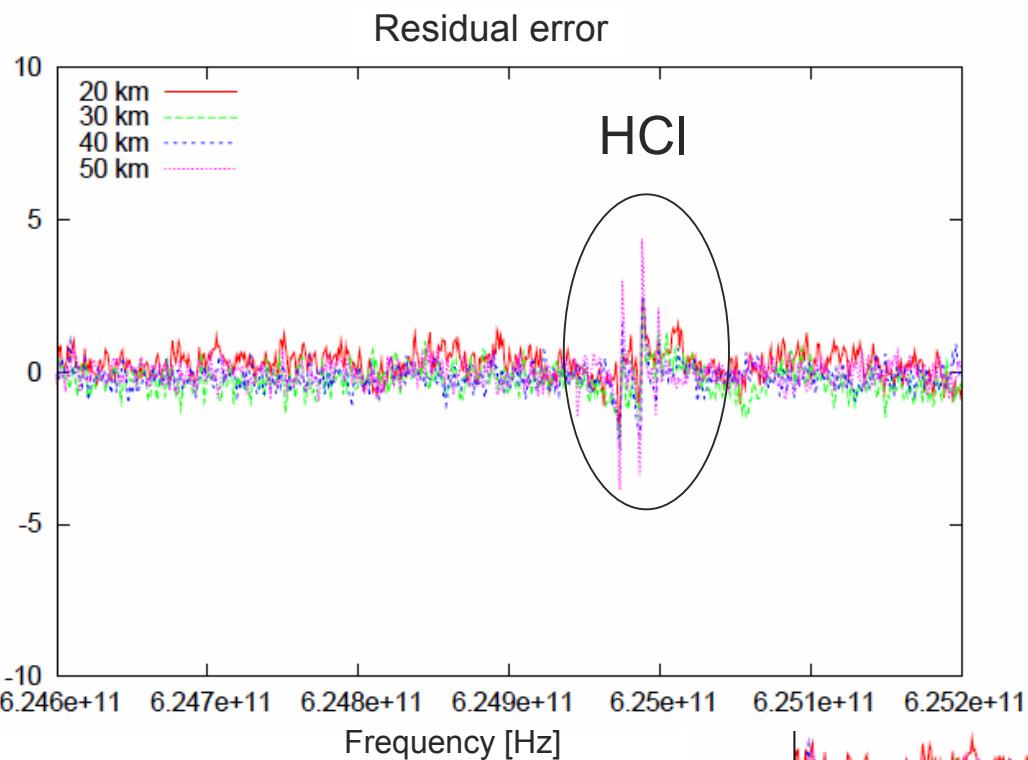
→ Compatible format with Aura/MLS



Retrieval Results



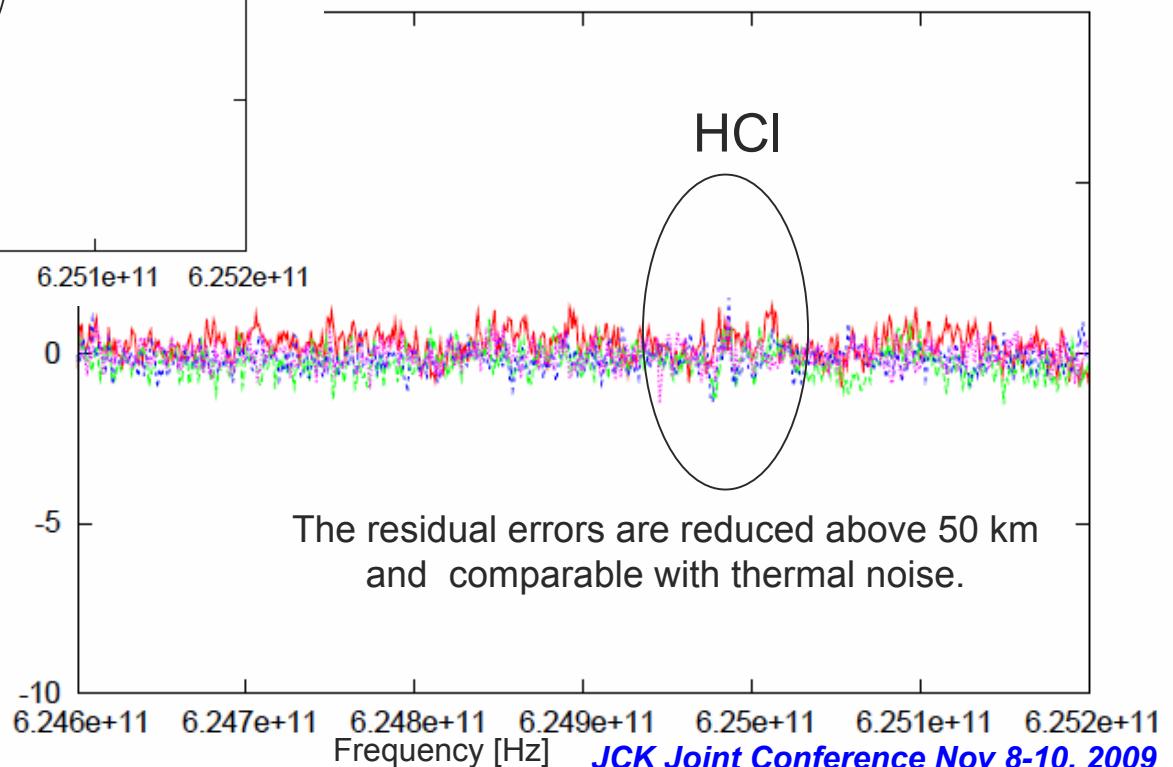
Doppler Shift by Earth's Rotation Rate



Before the Correction

After the Correction

Residual error



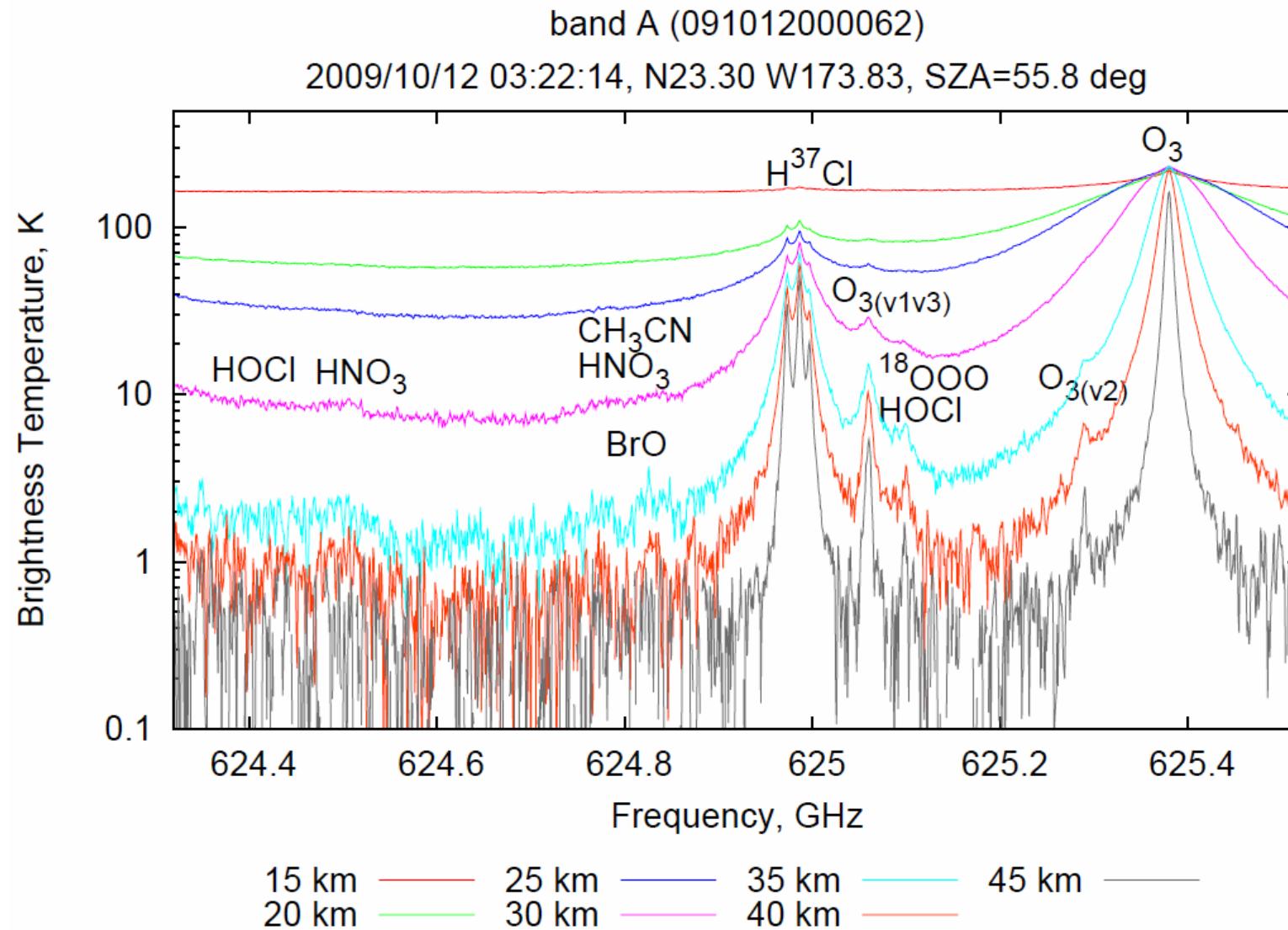
Algorithm for Noisy Product

- To avoid the bias from a priori, we retrieve the multi-scan data simultaneously [Livesey,2004]. i.e. The observation data \mathbf{y}_i ($i=1 \sim N$), the weighting function \mathbf{K}_i , the reference spectra \mathbf{f}_i , and the covariance matrix of the measurements \mathbf{S}_{yi} are represented by :

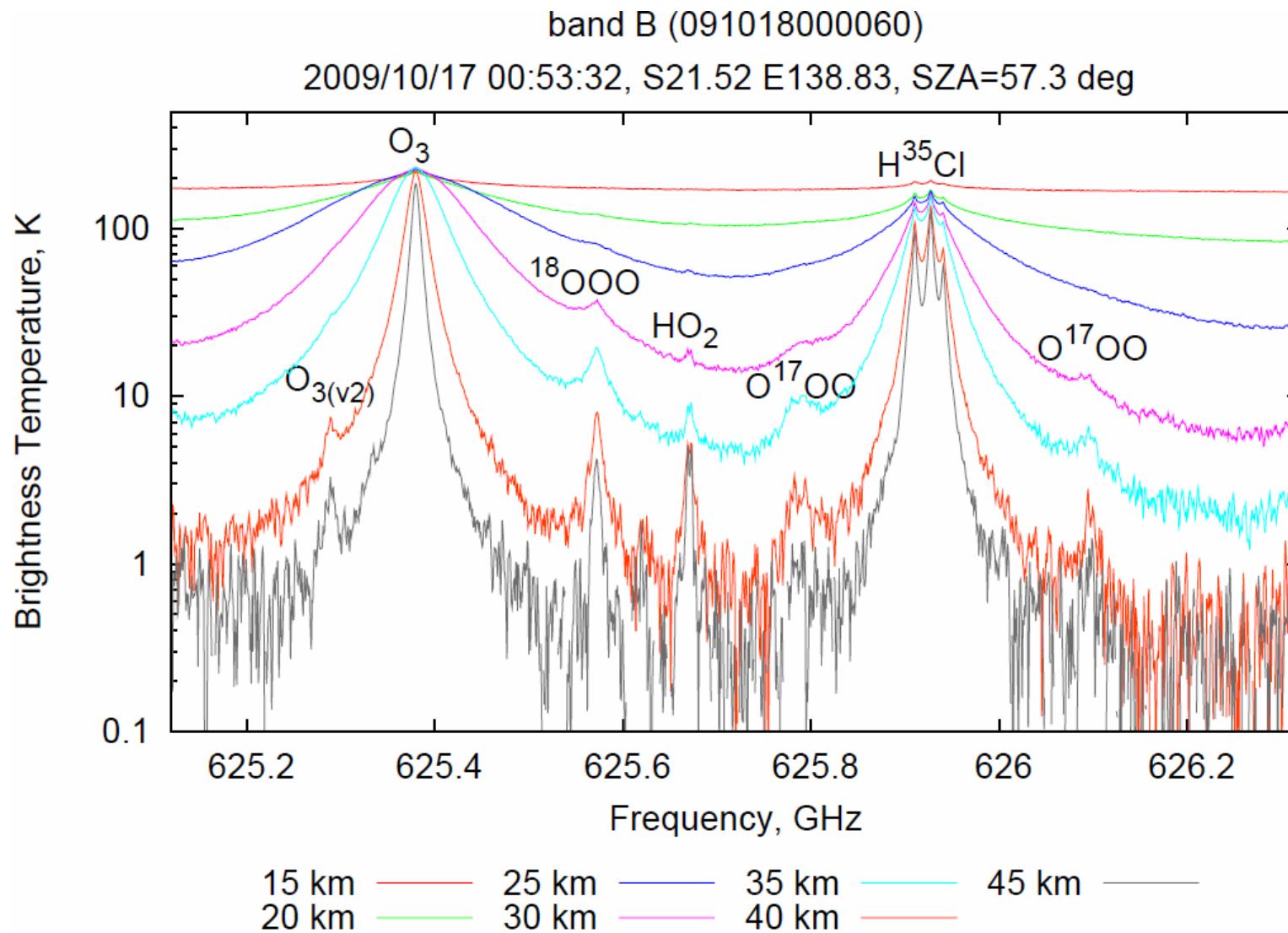
$$\mathbf{y} = \begin{pmatrix} \mathbf{y}_1 \\ \mathbf{y}_2 \\ \vdots \\ \mathbf{y}_N \end{pmatrix}, \quad \mathbf{f} = \begin{pmatrix} \mathbf{f}_{i1} \\ \mathbf{f}_2 \\ \vdots \\ \mathbf{f}_N \end{pmatrix}, \quad \mathbf{K} = \begin{pmatrix} \mathbf{K}_1 \\ \mathbf{K}_2 \\ \vdots \\ \mathbf{K}_N \end{pmatrix}, \quad \mathbf{S}_y = \begin{pmatrix} \mathbf{S}_{y1} & \mathbf{0} & \cdots & \mathbf{0} \\ \mathbf{0} & \mathbf{S}_{y1} & \mathbf{0} & \vdots \\ \vdots & \mathbf{0} & \ddots & \mathbf{0} \\ \mathbf{0} & \cdots & \mathbf{0} & \mathbf{S}_{y1} \end{pmatrix}$$

- To reduce the load of the system, we calculate below matrices and vectors, $\mathbf{K}_i^T \mathbf{S}_{yi} \mathbf{K}_i$ and $\mathbf{K}_i^T \mathbf{S}_{yi} (\mathbf{y}_i - \mathbf{f}_i)$ for each scan and save, because the size of these matrices and vectors are small.

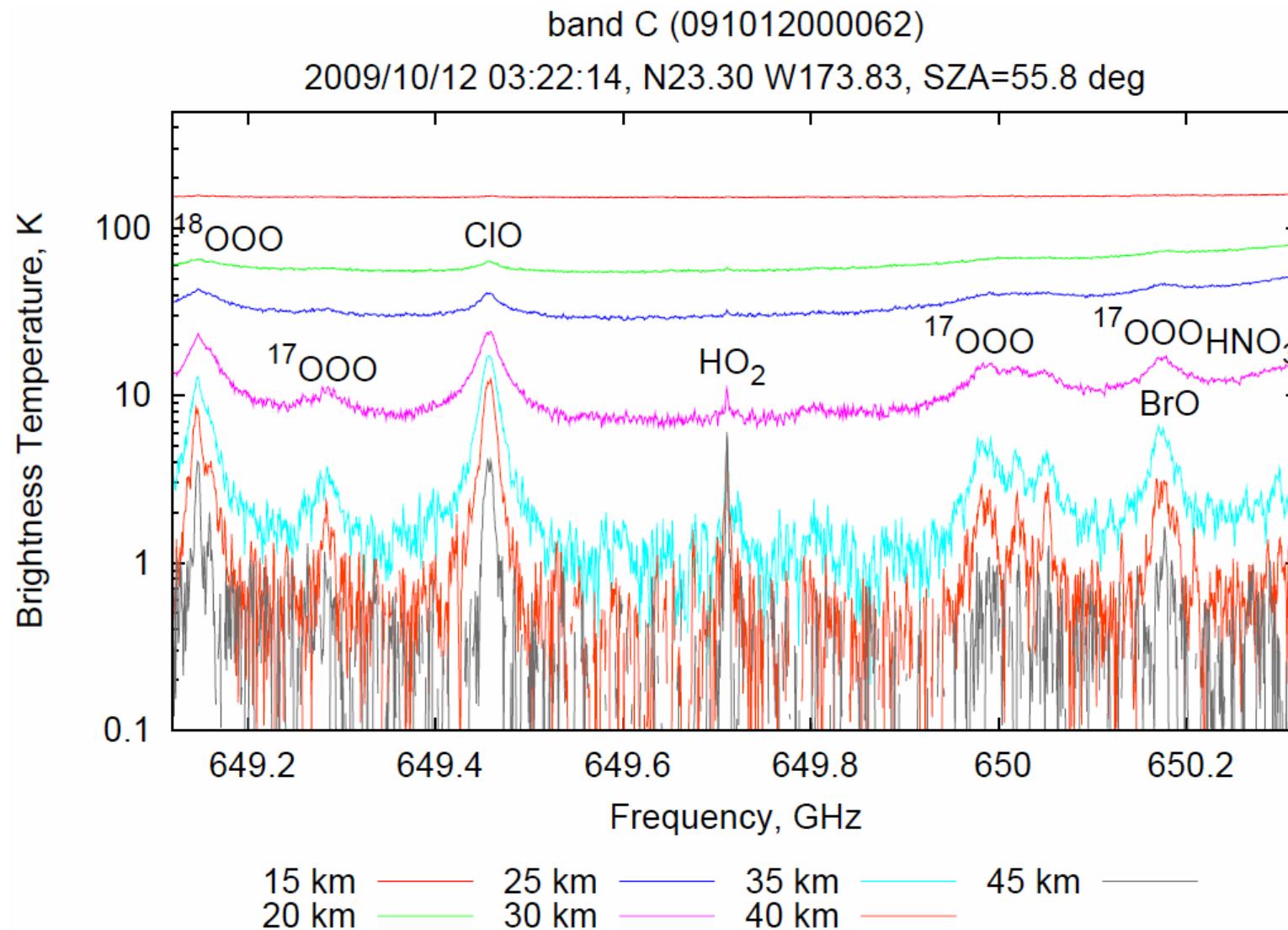
SMILES Level 1B Data – Band A



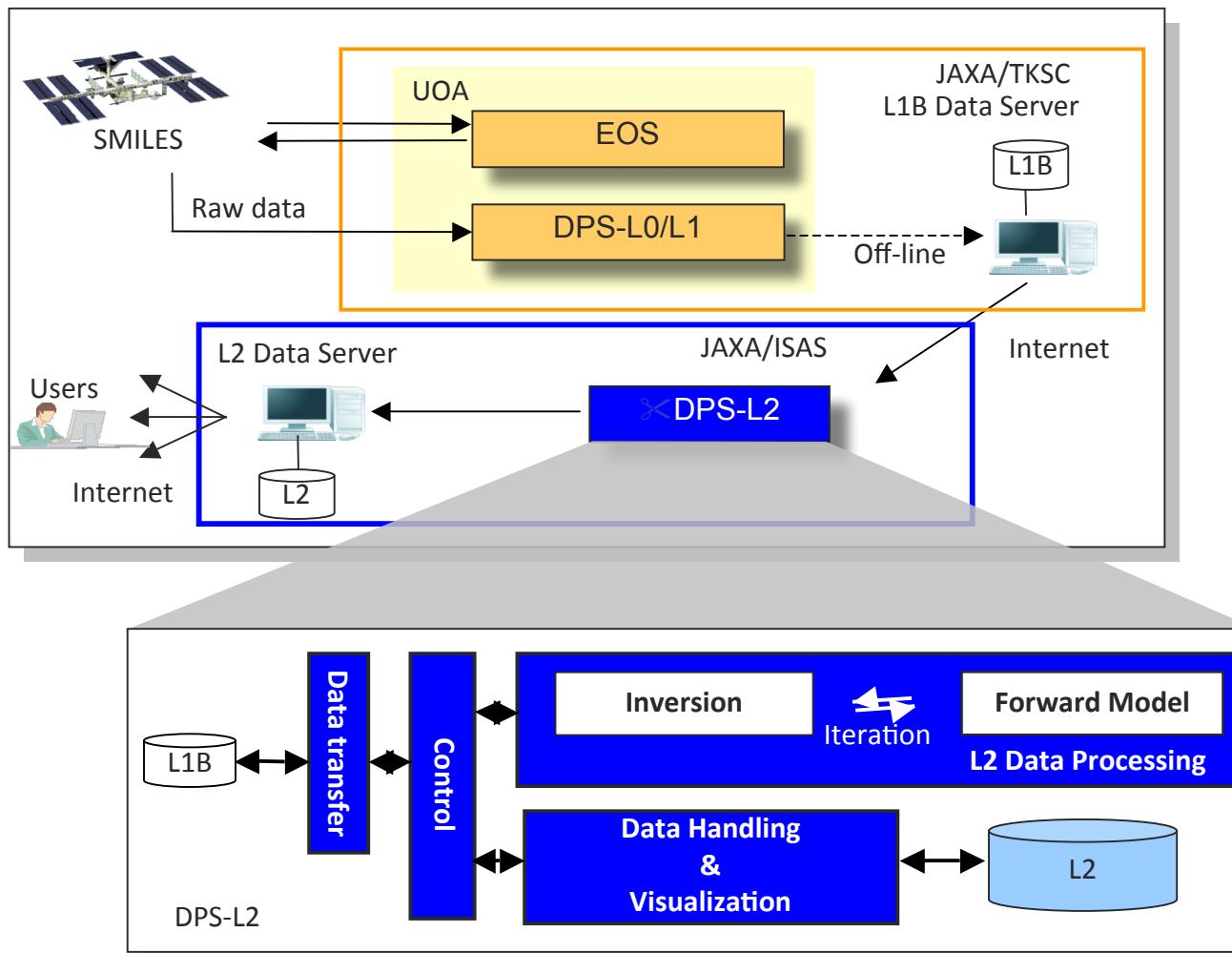
SMILES Level 1B Data – Band B



SMILES Level 1B Data – Band C



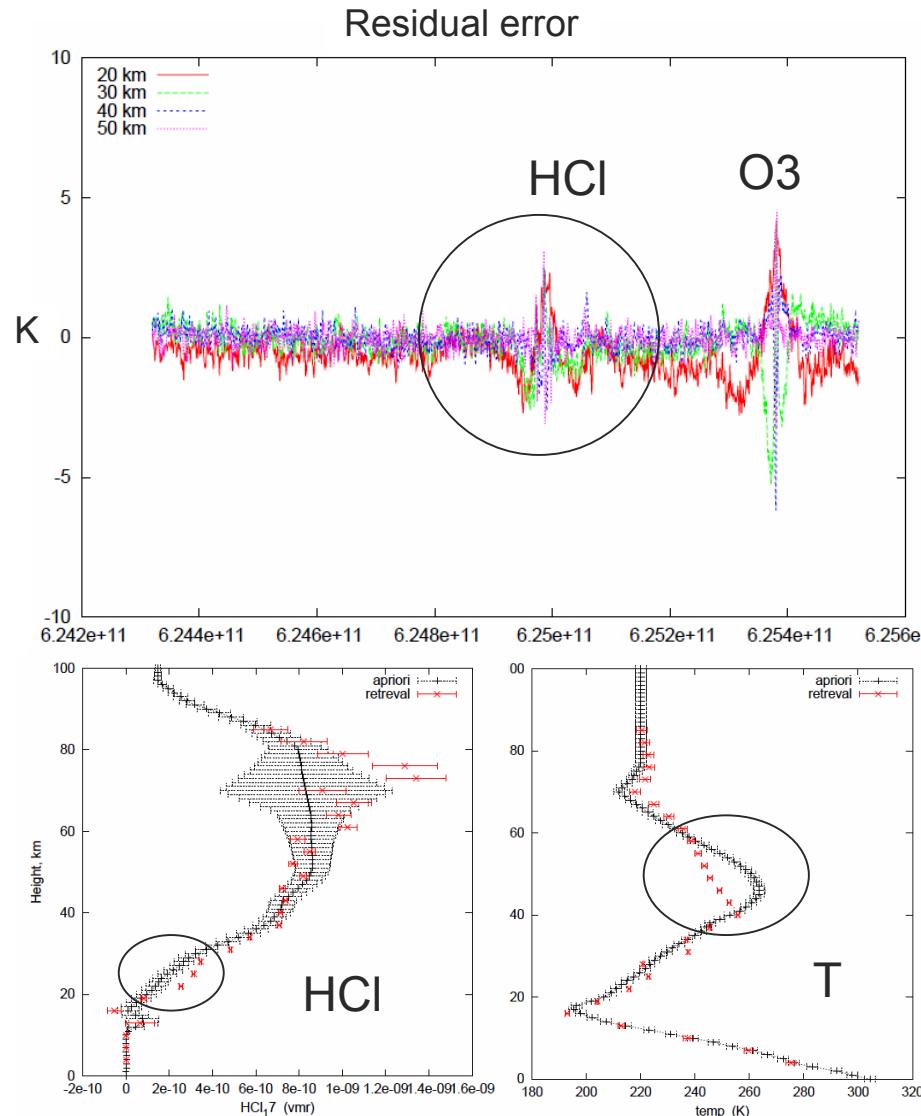
JEM/SMILES Data Flow



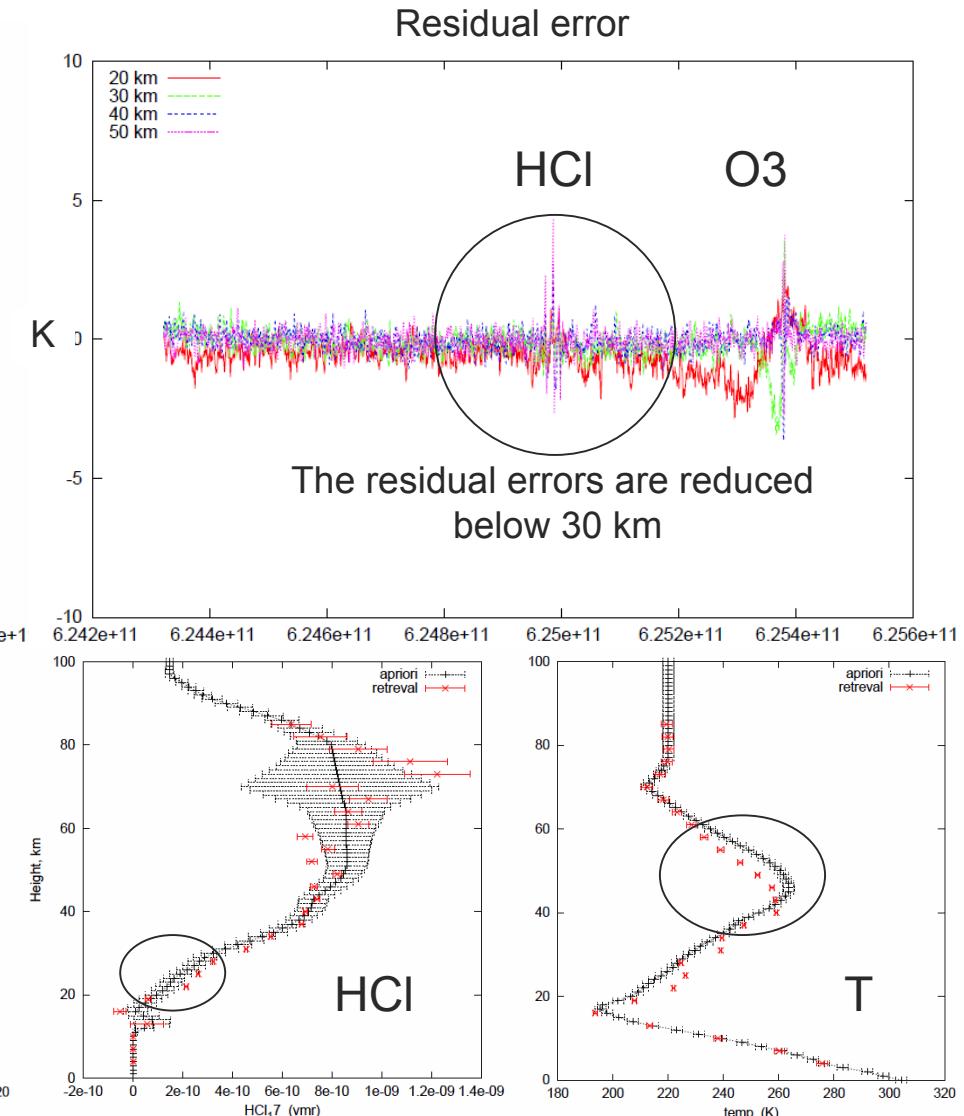
- Downlinked raw data from the SMILES will be received by the DPS-L0/L1 at User Operation Area (UOA) on Tsukuba Space Center (TKSC).
- The DPS-L0/L1 processes the raw data consisting of house keeping (HK) data and mission data to brightness temperature (level 1B data) in near-real-time.
- The DPS-L2 produces the vertical profiles of target species called level 2 data in near real time and distributes the level 2 data to data users by a Web server.

Pressure Broadening Parameters

Before the Correction

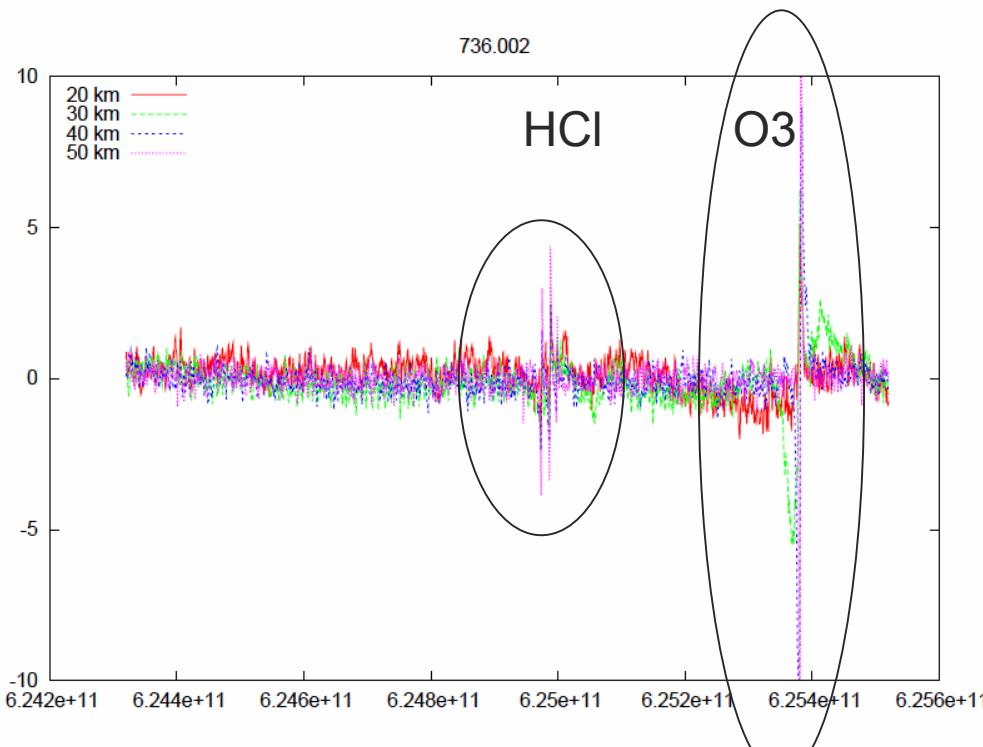


After the Correction



Earth's Rotation Rate

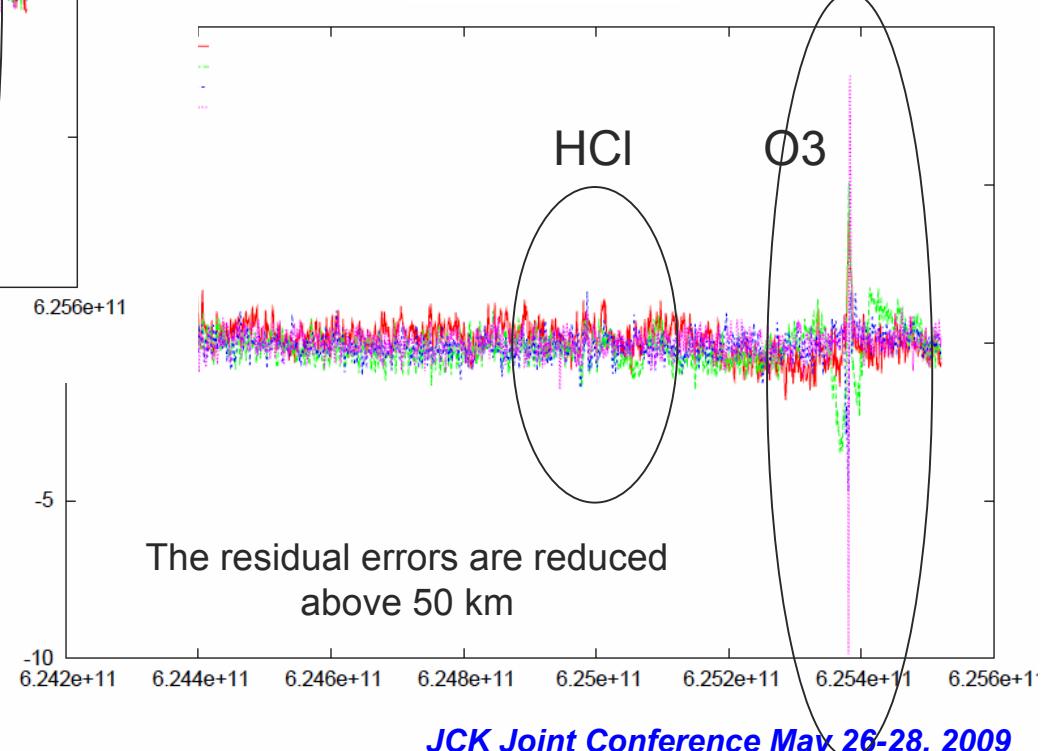
Residual error



Before the Correction

After the Correction

Residual error



Doppler shift & Instrument Functions

Doppler shift

- Velocity of the ISS : 8 km/s
- Rotation of the earth : 460 m/s (on the equator)
- Wind : ≤ 100 m/s.

FOV Convolution :

Considering the effects from tilt of ISS

$$T_A(\nu, z_0) = \int_{z_{min}}^{z_{max}} P(z, z_0) \cdot T_p(\nu, z) dz,$$

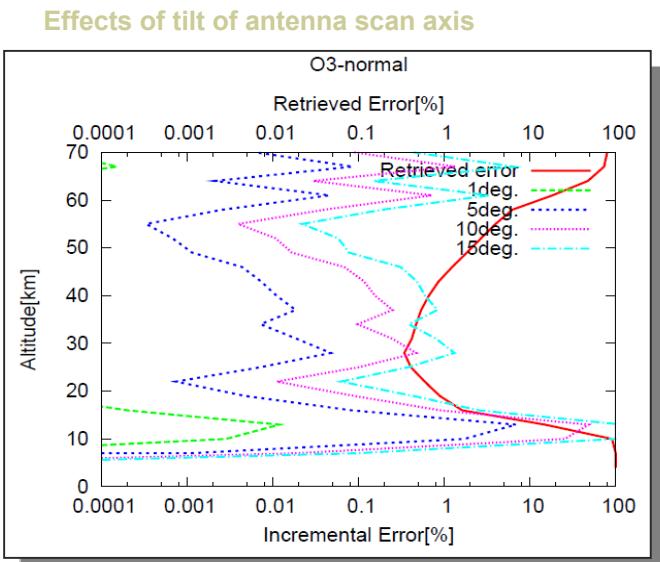
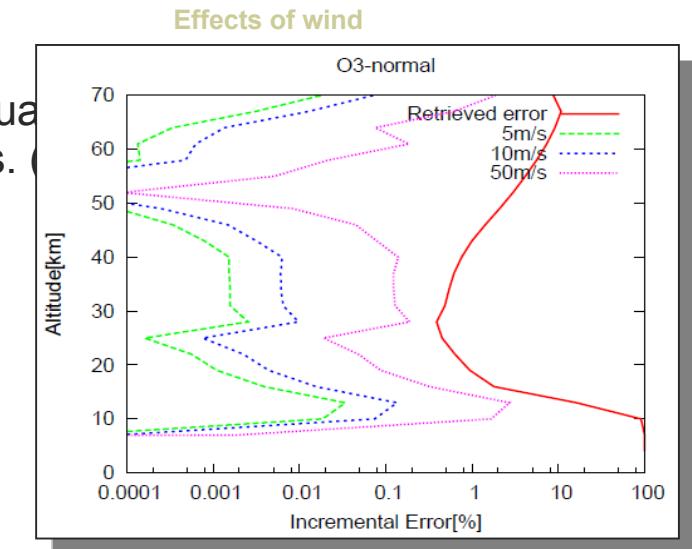
Fold Sidebands (Single Sideband Separator)

$$\begin{aligned} T_{mix:i}(\nu, z_0) &= \begin{bmatrix} K_{i,a}^{LSB}(\nu_{LO} - \nu_{if}, z_0) \\ K_{i,c}^{LSB}(\nu_{LO} - \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} - \nu_{if}, z_0) \\ T_c(\nu_{LO} - \nu_{if}) \end{bmatrix} \\ &+ \begin{bmatrix} K_{i,a}^{USB}(\nu_{LO} + \nu_{if}) \\ K_{i,c}^{USB}(\nu_{LO} + \nu_{if}) \end{bmatrix}^T \cdot \begin{bmatrix} T_A(\nu_{LO} + \nu_{if}, z_0) \\ T_c(\nu_{LO} + \nu_{if}) \end{bmatrix} \\ K_{i,j}^{LSB,USB}(\nu, T) &= \frac{1 + \alpha(T)^2 + 2\alpha(T)\cos\left(\frac{m\pi\nu}{\nu_0(T)}\right)}{4}. \end{aligned}$$

Channel Average: Considering channel dependence

$$T_{AOS(l)}(\nu_j, z_0) = \frac{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) \cdot T_{mix(k)}(\nu, z_0) d\nu}{\int_{\nu_{min}}^{\nu_{max}} H_{AOS(l)}(\nu - \nu_j) d\nu}.$$

$$H_{AOS(l)}(\nu - \nu_j) = \sum_{j=1}^{N_j} \frac{A_{i,j}}{w_{i,j}\sqrt{\pi/2}} \cdot \exp\left(-2\frac{(\nu - \nu_j - xc_{i,j})^2}{w_{i,j}^2}\right).$$

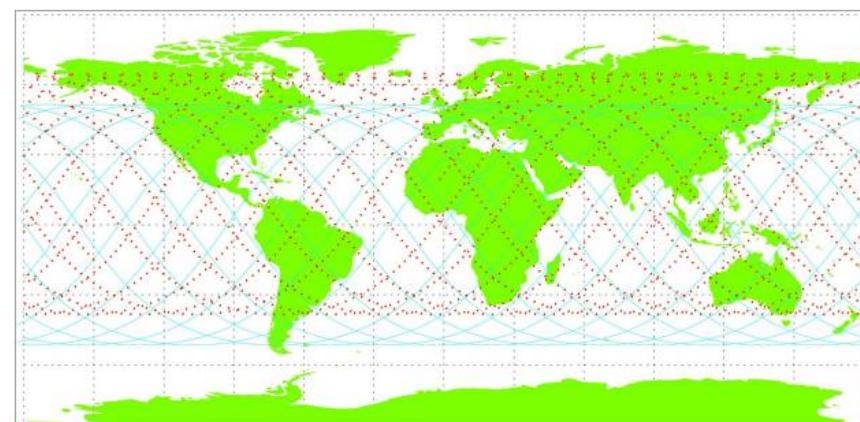
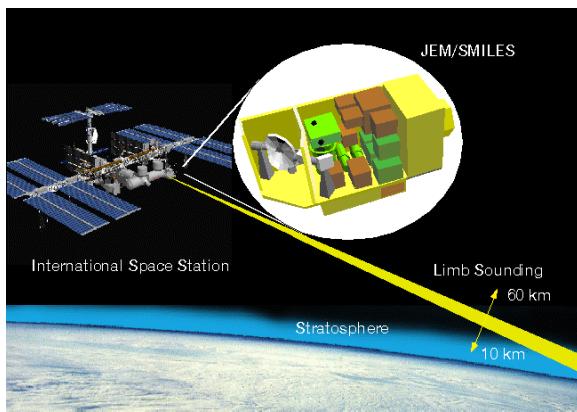


Outline of JEM/SMILES

- The Superconducting Submillimeter-wave Limb-Emission Sounder (SMILES) had been launched and aboard the Japanese Experiment Module (JEM) of the International Space Station (ISS) in **Sep, 2009**.
- The SMILES carries 4K cooled **Superconductor-Insulator-Superconductor (SIS) mixers** to demonstrate a highly sensitive instrument for submillimeter limb-emission sounding.
 - Nominal latitude coverage: **38S - 65N**
 - Observation number : 1600 / day (105 observation / orbit, 15.5 orbit / day)
 - Observation band number: 3 (band A, B, C)

Standard products

O₃, HCl, ClO, HNO₃, CH₃CN, HO₂, HOCl, BrO, O₃ isotopes
(in the middle atmosphere)



Contents

- Outline of the SMILES Level 2 data processing system (DPS-L2) in ISAS/JAXA
- Retrieval algorithm of the DPS-L2
- Study of the observation capability for target species.
- Improvements of the data quality by using Level 1B data (calibrated spectra)

- Recent status of the Level 2 products
- Distribution system of the Level 2 data

Etc.

● S6-05 Masato SHIOTANI*, Masahiro TAKAYANAGI and JEM/SMILES MISSION TEAM:
CURRENT STATUS OF SUPERCONDUCTING SUBMILLIMETER-WAVE LIMB-EMISSION
SOUNDER (SMILES)

● S6-06 Chikako TAKAHASHI*, Chihiro MITSUDA, Makoto SUZUKI, Yoshitaka IWATA, Hiroo
HAYASHI, Koji IMAI, et al.: CAPABILITY STUDY FOR ATMOSPHERIC MINOR SPECIES WITH
JEM/SMILES

● S6-07 Yasuko KASAI*, Philippe BARON, Jana MENDROK, Satoshi OCHIAI, Takeshi
MANABE, et al.: OBSERVATION CAPABILITIES OF SUPERCONDUCTIVE SUBMILLIMETER-
WAVE LIMB-EMISSION SOUNDER (SMILES) ONBOARD INTERNATIONAL SPACE STATION

● S6-08 Chihiro MITSUDA*, Chikako TAKAHASHI, Makoto SUZUKI, Hiroo HAYASHI, Takuki
SANO, et al.: DEVELOPMENT OF JEM/SMILES LEVEL 2 DATA PROCESSING SYSTEM